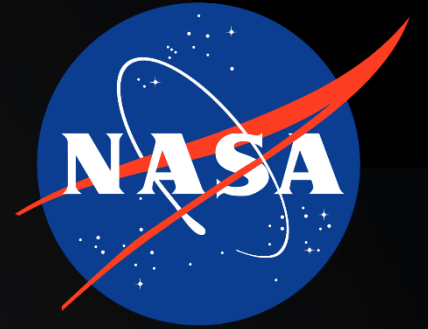


National Aeronautics and Space Administration



# Advanced Manufacturing for Propulsion Systems: Additive Manufacturing, Optical Measurement, and Composites

**Paul Gradl**

National Aeronautics and Space Administration (NASA)

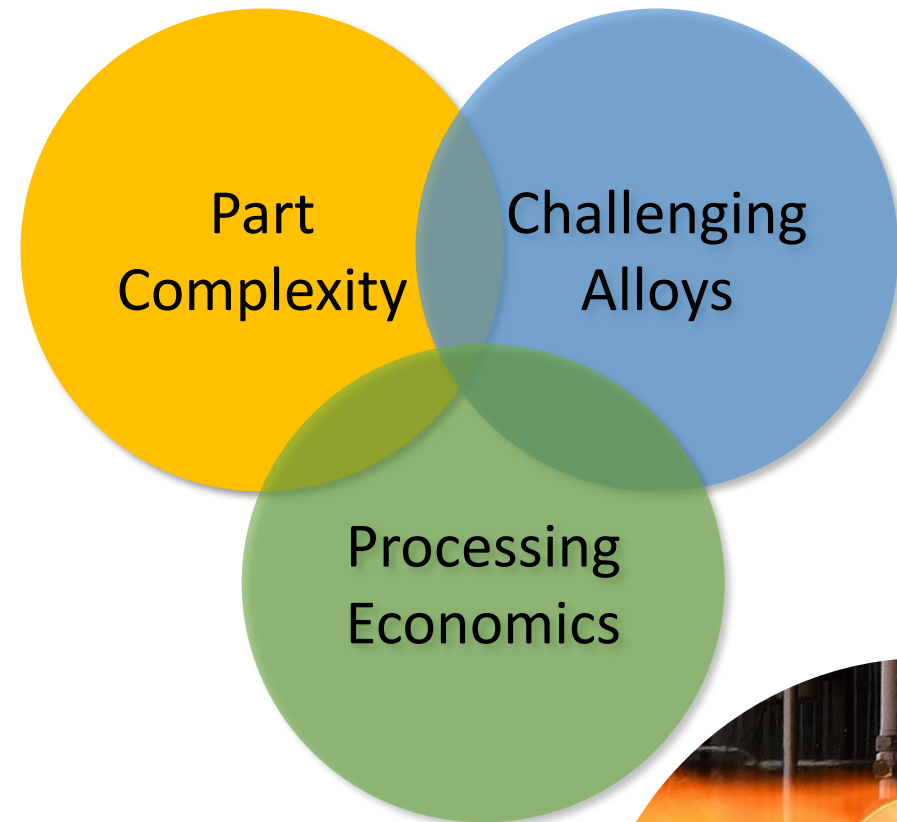
27 March 2023



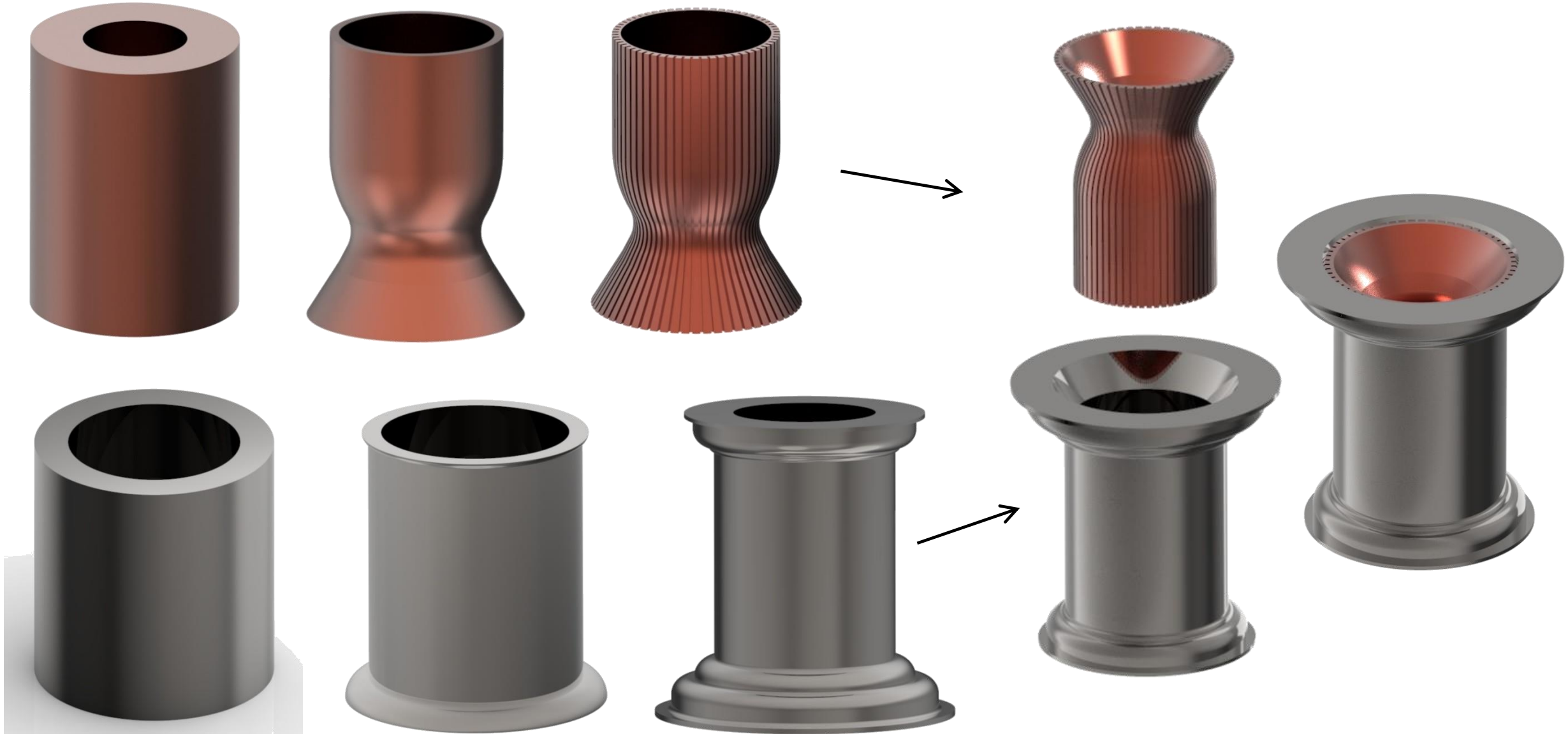
# The Case for Additive Manufacturing in Propulsion



- Metal Additive Manufacturing (AM) can provide significant advantages for lead time and cost over traditional manufacturing for rocket engines.
  - Lead times reduced by 2-10x
  - Cost reduced by more than 50%
- Complexity is inherent in liquid rocket engines and AM provides new design and performance opportunities.
- Materials that are difficult to process using traditional techniques, long-lead, or not previously possible are now accessible using metal additive manufacturing.

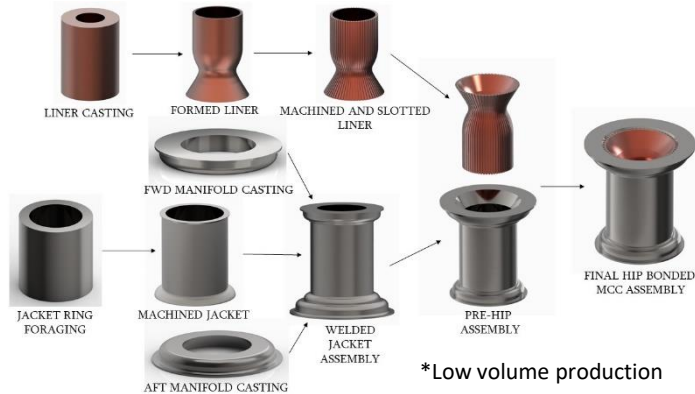


# Traditional Manufacturing...Forging to final assembly





# A rocket combustion chamber case study for AM

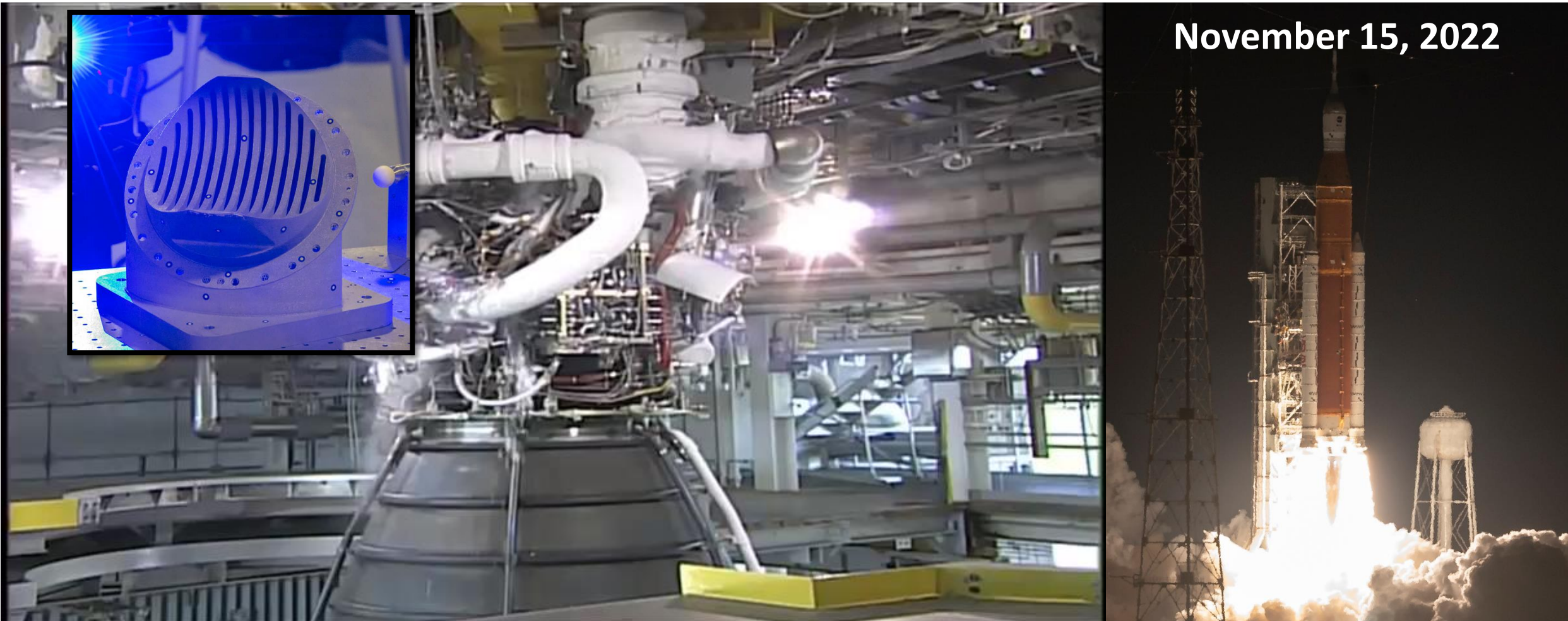
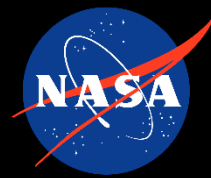


Category	Traditional Manufacturing	Initial AM Development	Evolving AM Development
<b>Design and Manufacturing Approach</b>	Multiple forgings, machining, slotting, and joining operations to complete a final multi-alloy chamber assembly	Four-piece assembly using multiple AM processes; limited by AM machine size. Two-piece L-PBF GRCo-84 liner and EBW-DED Inconel 625 jacket	Three-piece assembly with AM machine size restrictions reduced and industrialized. Multi-alloy processing; one-piece L-PBF GRCo-42 liner and Inconel 625 LP-DED jacket
<b>Schedule (Reduction)</b>	18 months	8 months (56%)	5 months (72%)
<b>Cost (Reduction)</b>	\$310,000	\$200,000 (35%)	\$125,000 (60%)

**As AM process technologies evolve using multi-materials and processes, additional design and programmatic advantages are being discovered**

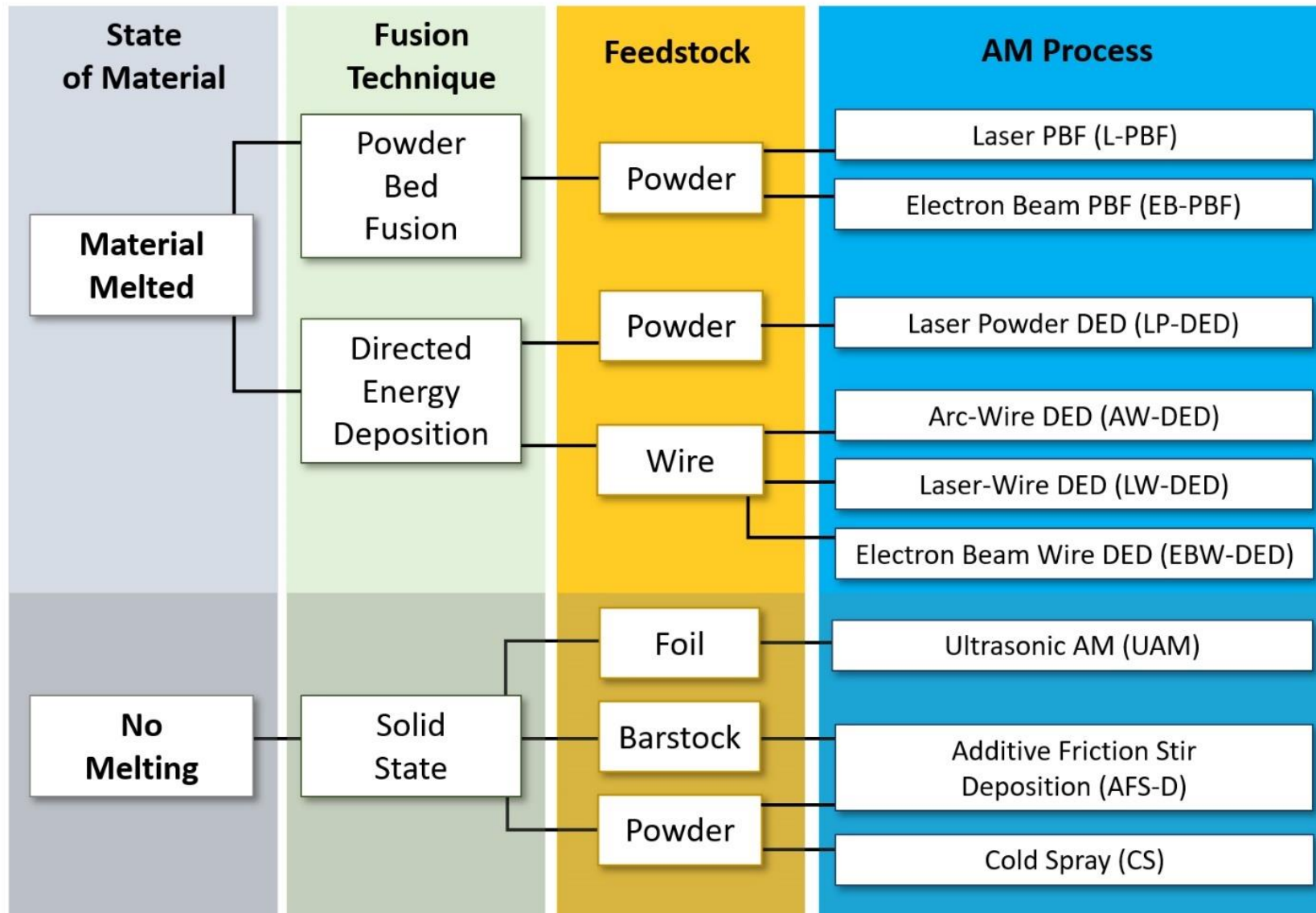


# Additive Manufacturing in use on NASA Space Launch System (SLS)



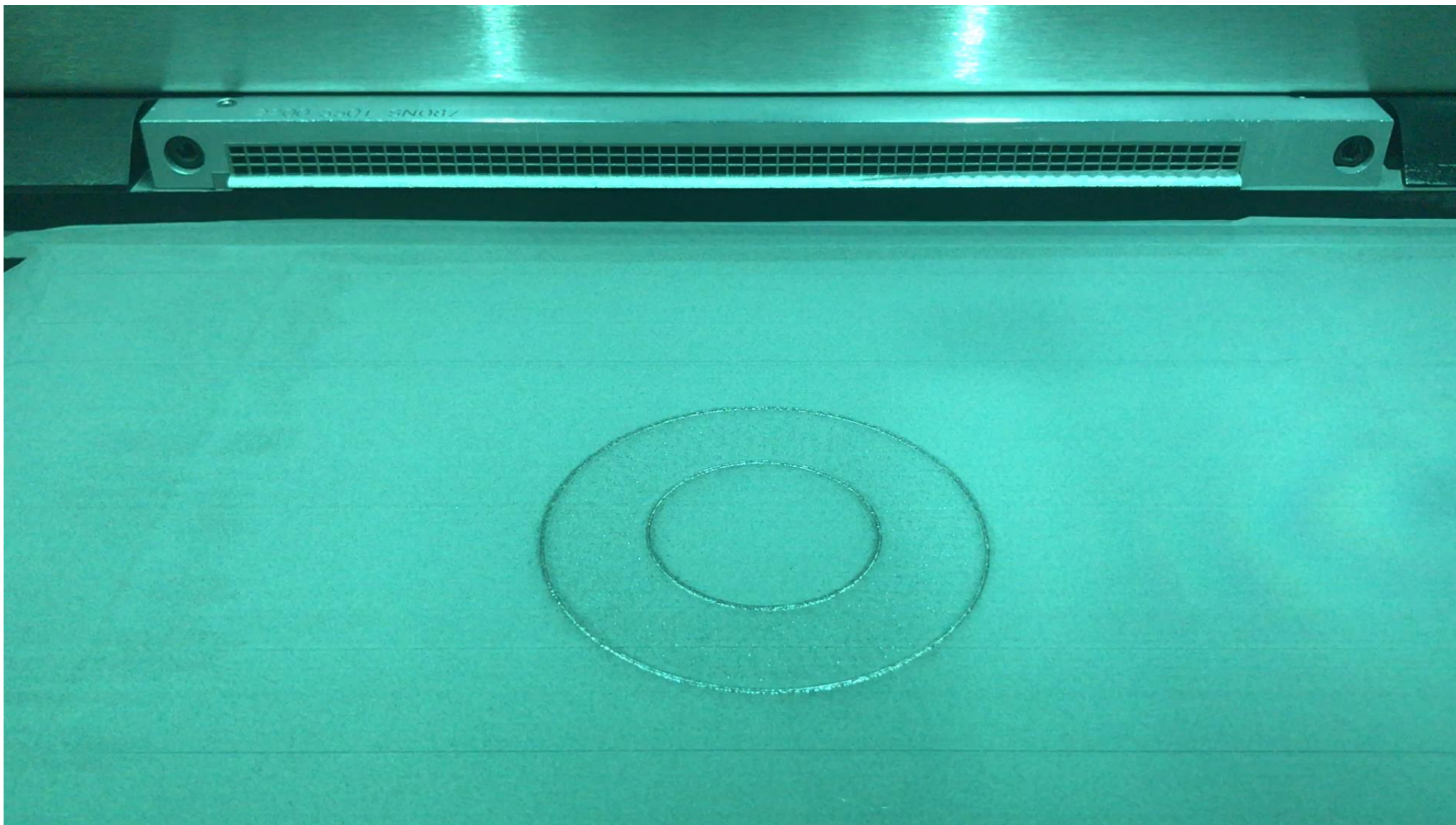
**Successful hot-fire testing of full-scale additive manufacturing (AM) Part to be flown on SLS RS-25  
RS-25 Pogo Z-Baffle – Used existing design with AM to reduce complexity from 127 welds to 4 welds**

# Various Metal AM Processes

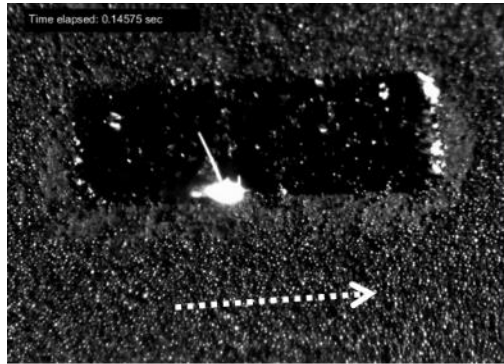




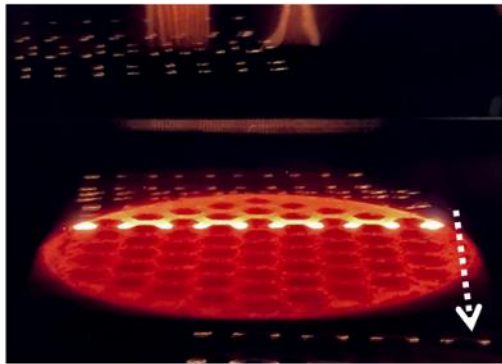
# Laser Powder Bed Fusion (L-PBF)



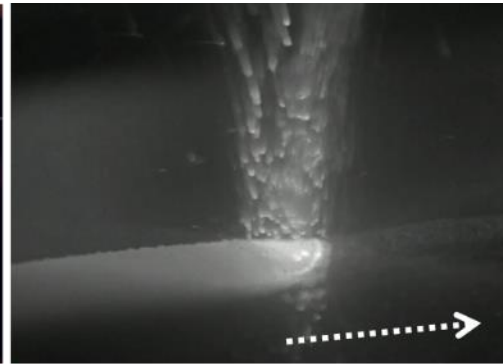
# AM Processes for various applications



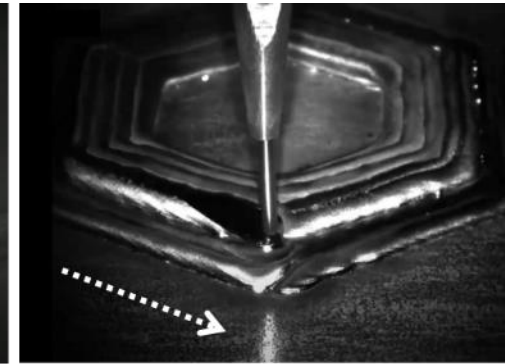
Laser Powder Bed Fusion



Electron Beam Powder Bed Fusion



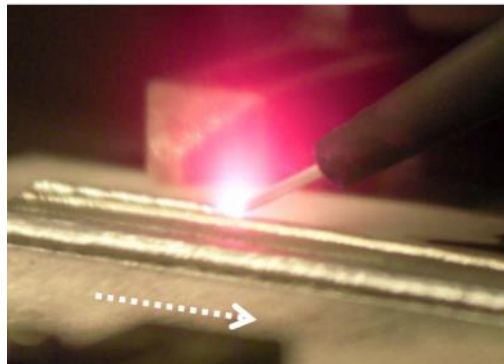
Laser Powder DED



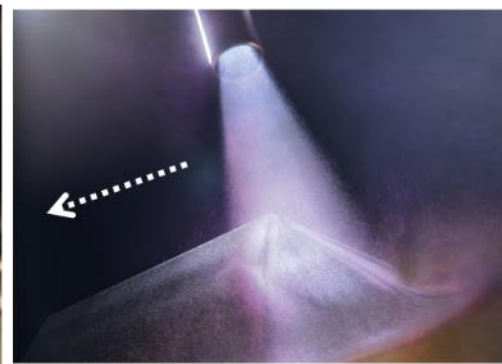
Laser Wire DED



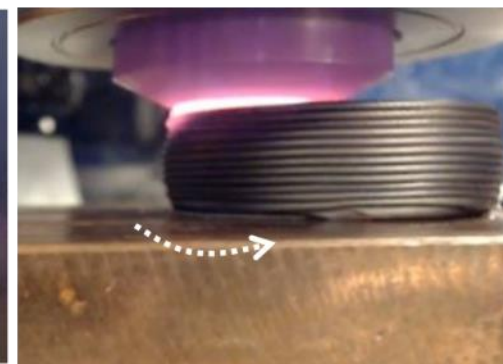
Arc Wire DED



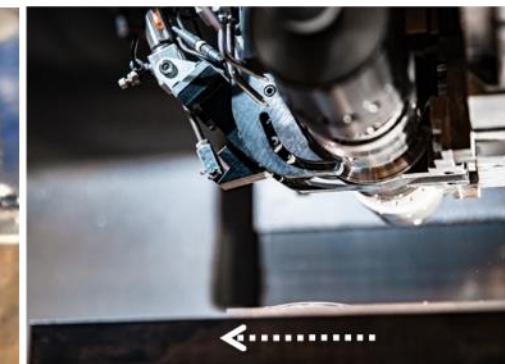
Electron Beam Wire DED



Cold Spray



Additive Friction Stir Deposition



Ultrasonic Additive Manufacturing

A) Laser Powder Bed Fusion [<https://doi.org/10.1016/j.actamat.2017.09.051>], B) Electron Beam Powder Bed Fusion [Credit: Courtesy of Freemelt AB, Sweden], C) Laser Powder DED [Credit: Formally], D) Laser Wire DED [Credit: Ramlab and Cavitar], E) Arc Wire DED [Credit: Institut Maupertuis and Cavitar], F) Electron Beam DED [NASA], G) Cold spray [Credit: LLNL], H) Additive Friction Stir Deposition [NASA], I) Ultrasonic AM [Credit: Fabrisonic].



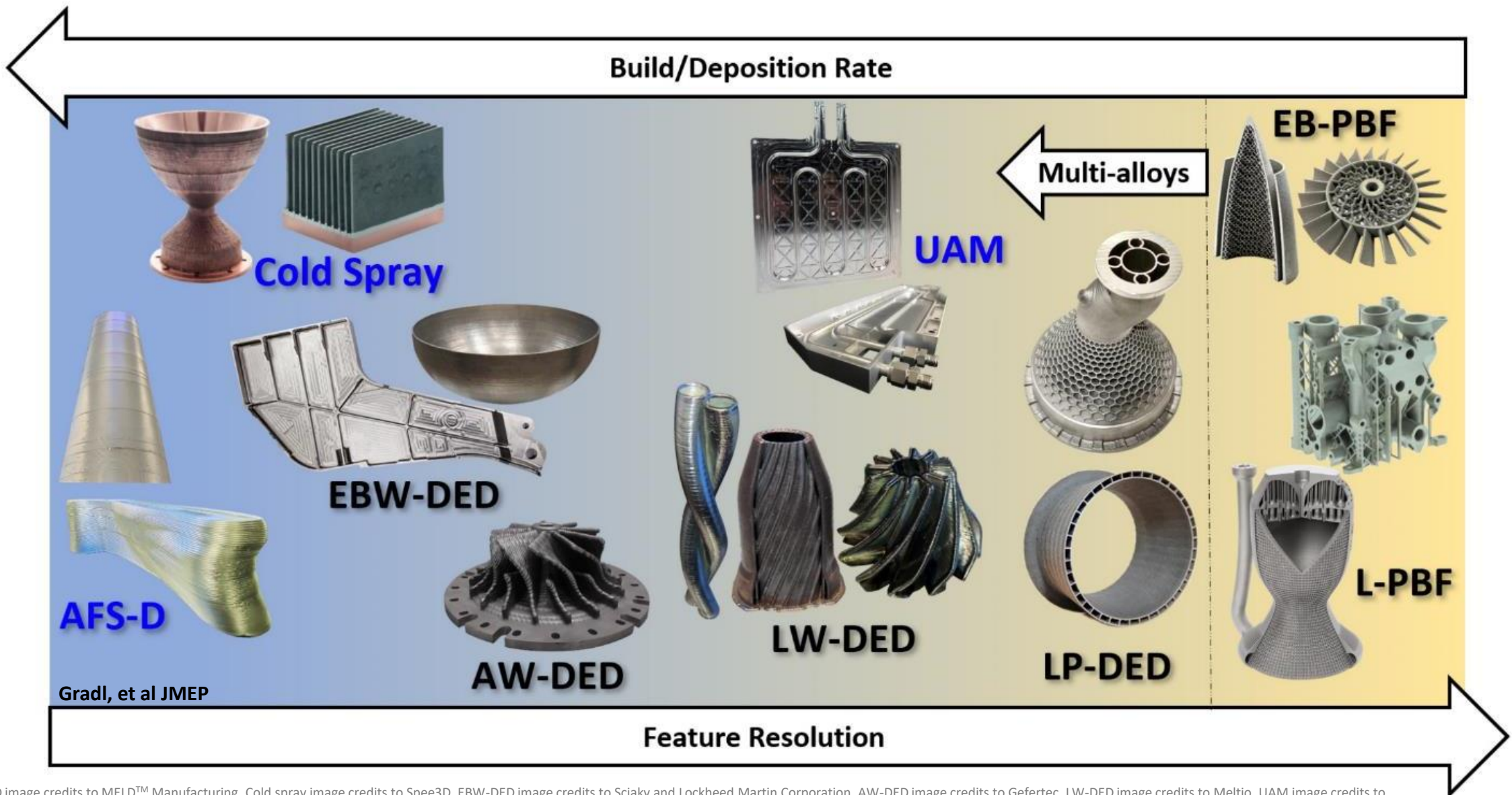
# Methodical AM Process Selection



- What is the **alloy** required for the application?
- What is the **overall part size**?
- What is the **feature resolution** and internal **complexities**?
- Is it a **single alloy** or **multiple**?
- What are **programmatic requirements** such as cost, schedule, risk tolerance?
- What are the end-use environments and **properties required**?
- What is the **qualification/certification** path for the application/process?



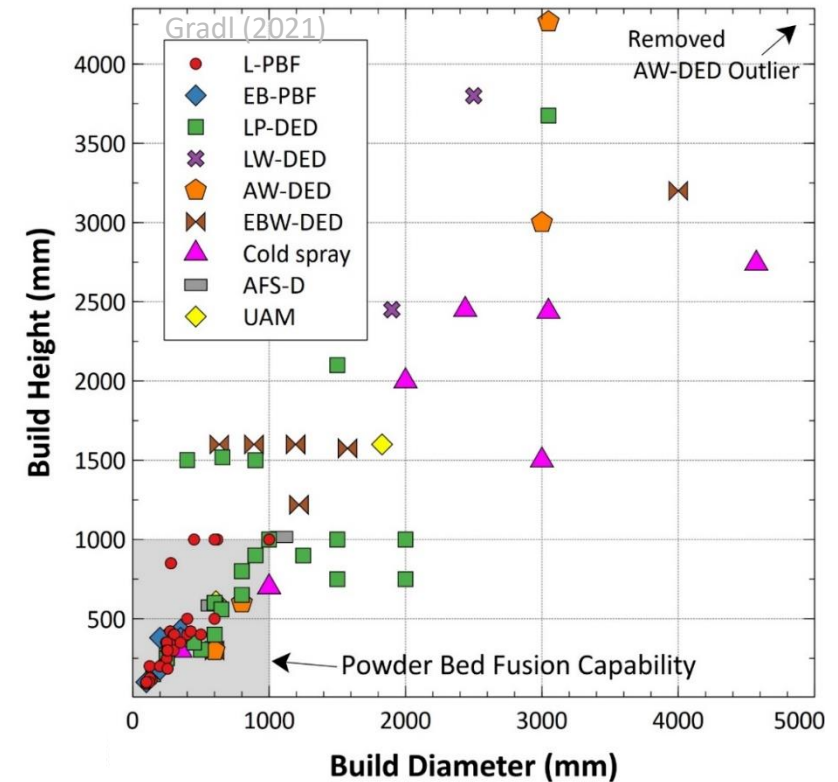
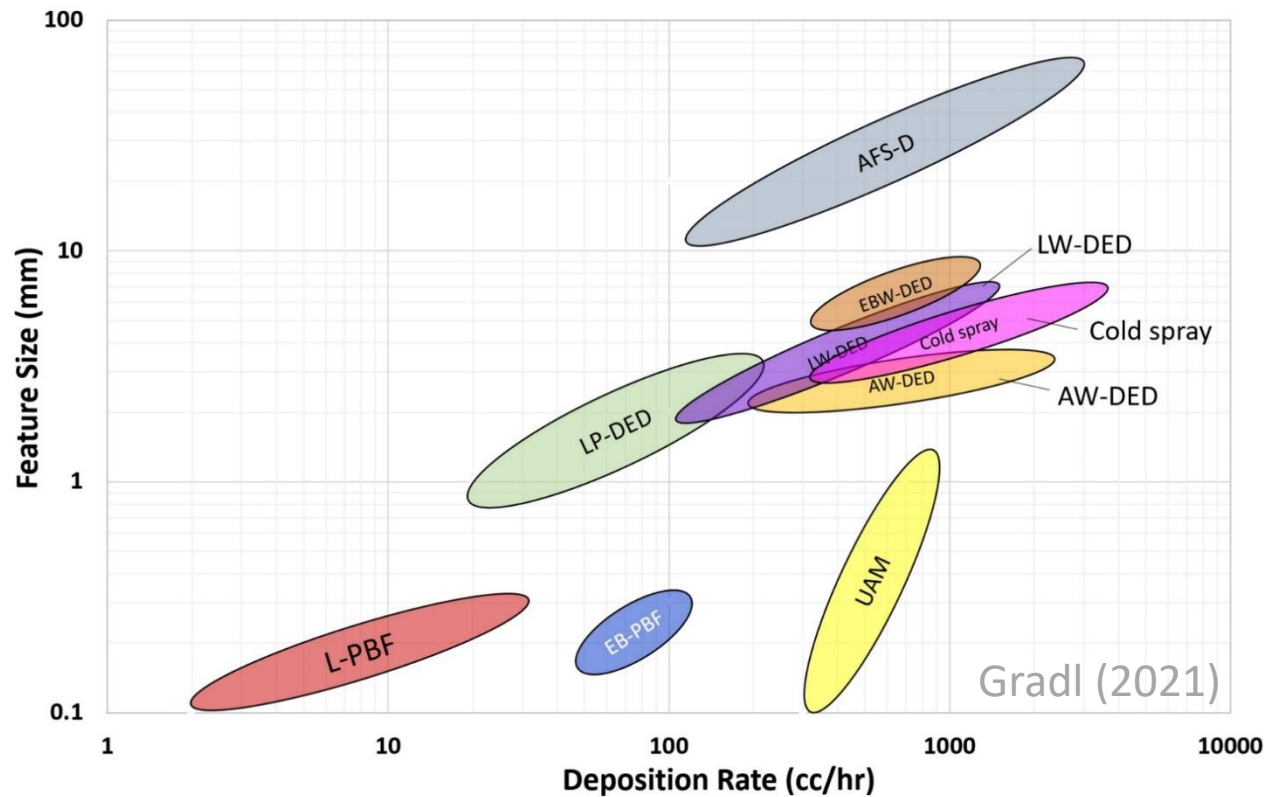
# Criteria and Comparison Various Metal AM Processes



**CREDITS:** AFS-D image credits to MELD™ Manufacturing, Cold spray image credits to Spee3D, EBW-DED image credits to Sciaky and Lockheed Martin Corporation, AW-DED image credits to Gefertec, LW-DED image credits to Meltio, UAM image credits to Fabrisonic and NASA JPL, LP-DED image credits to DEPOZ project led by IRT Saint-Exupery and Formally, L-PBF image credits to Renishaw plc and CellCore GmbH/Sol Solutions Group AG, EB-PBF image credits to Wayland and GE Additive/Arcom.



# Various criteria for selecting AM techniques



Complexity of Features

Scale of Hardware

Material Physics

Cost

Material Efficiency

Speed of Process

Material Properties

Internal Geometry

Availability

Post Processing





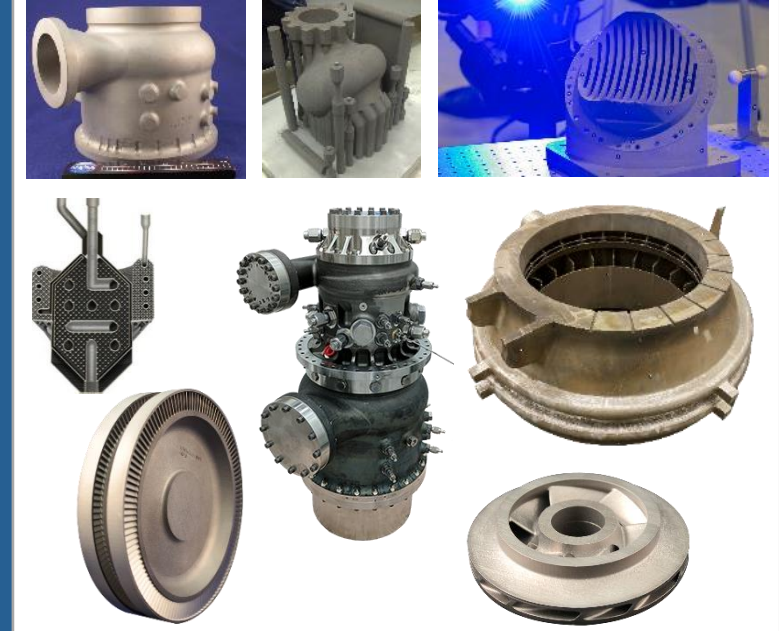
# Additive Manufacturing (AM) Development at NASA for Liquid Rocket Engines



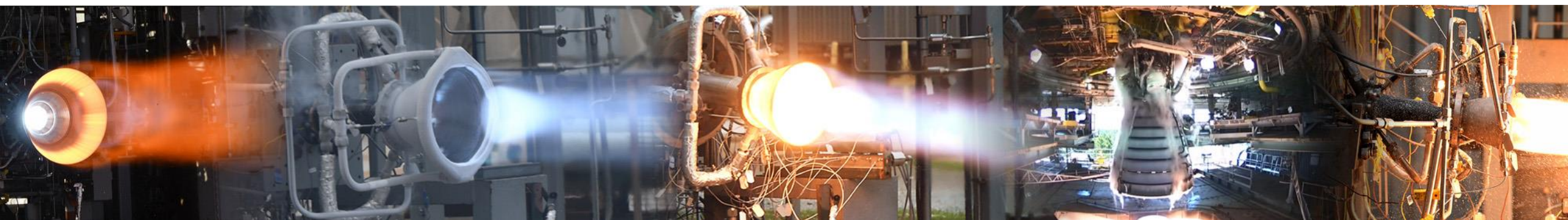
Laser Powder Bed Fusion (L-PBF)  
Copper Alloys combined with other  
AM processes to provide bimetallic



Directed Energy Deposition

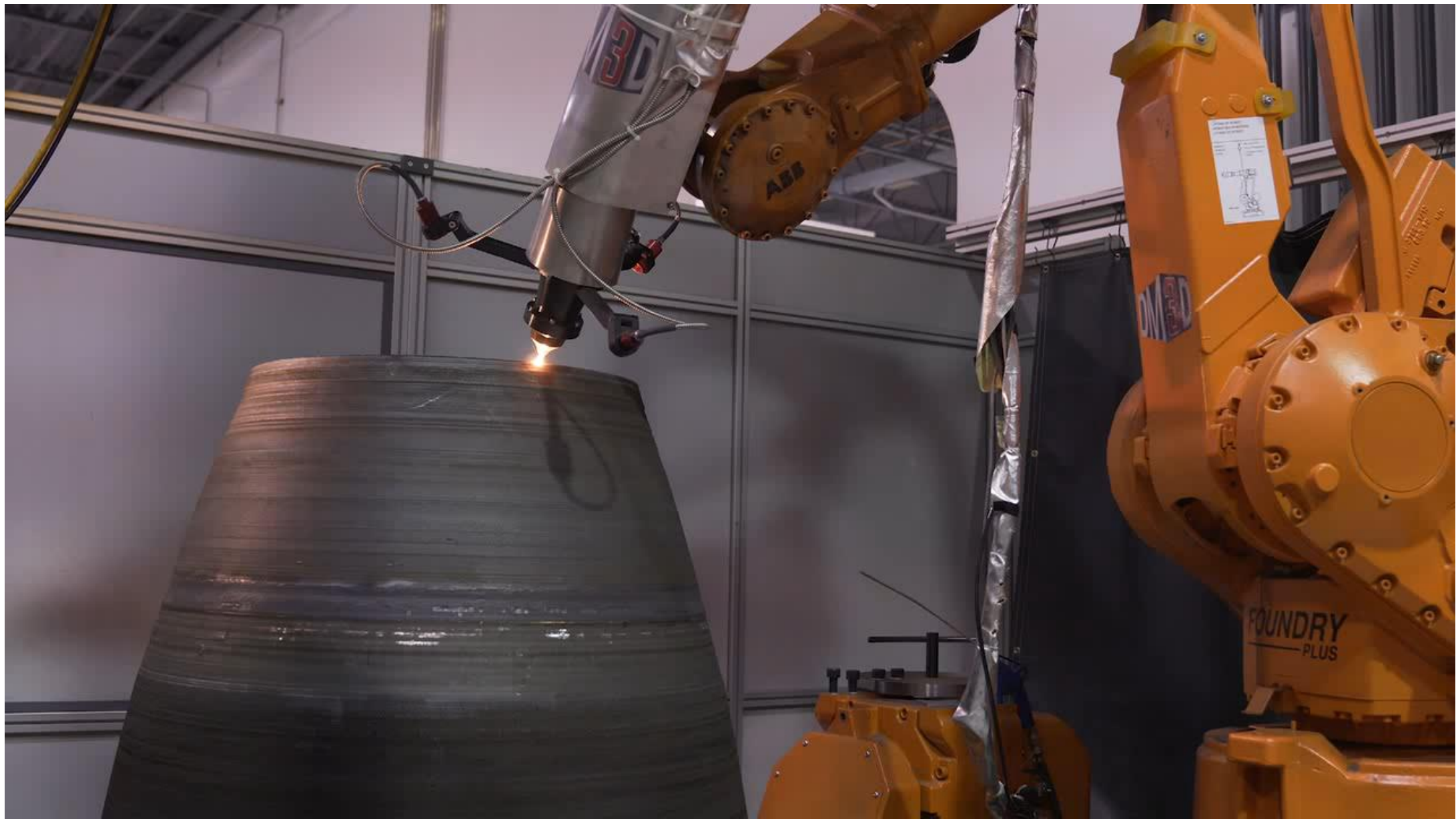


L-PBF of complex components, new  
alloy developments for harsh  
environment

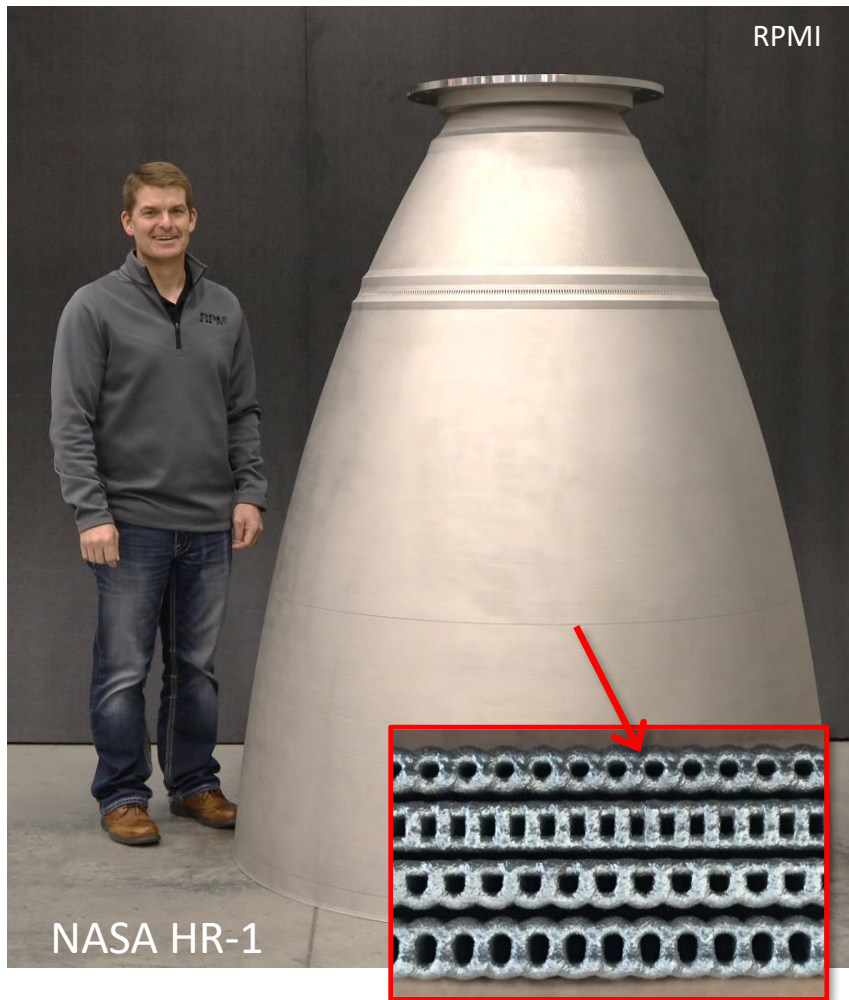




# Laser Powder Directed Energy Deposition (DED)



# Laser Powder Directed Energy Deposition (LP-DED) Large Scale Nozzles



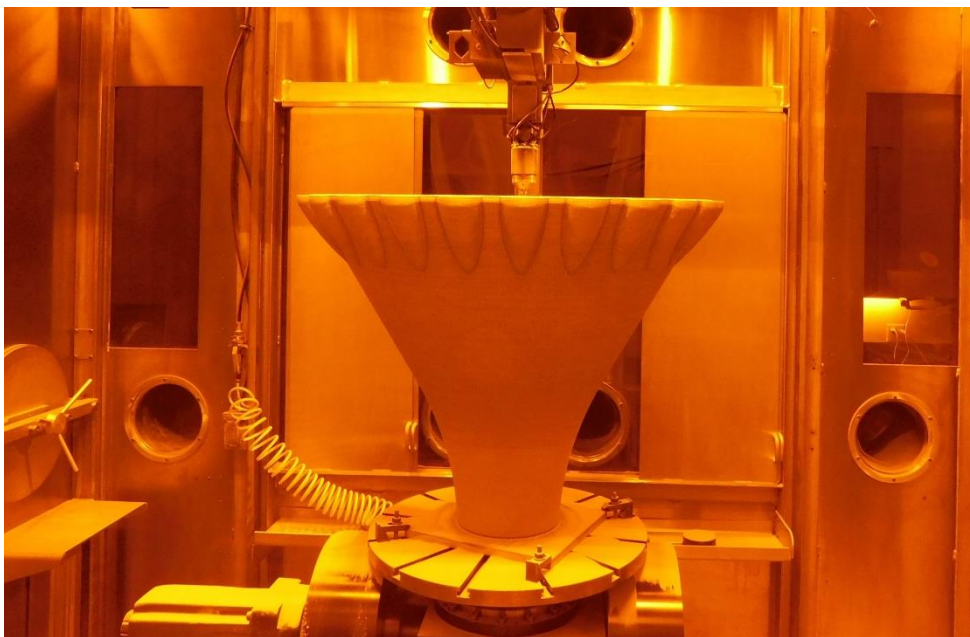
**60" (1.52 m) diameter and 70" (1.78 m) height with integral channels**  
90 day deposition



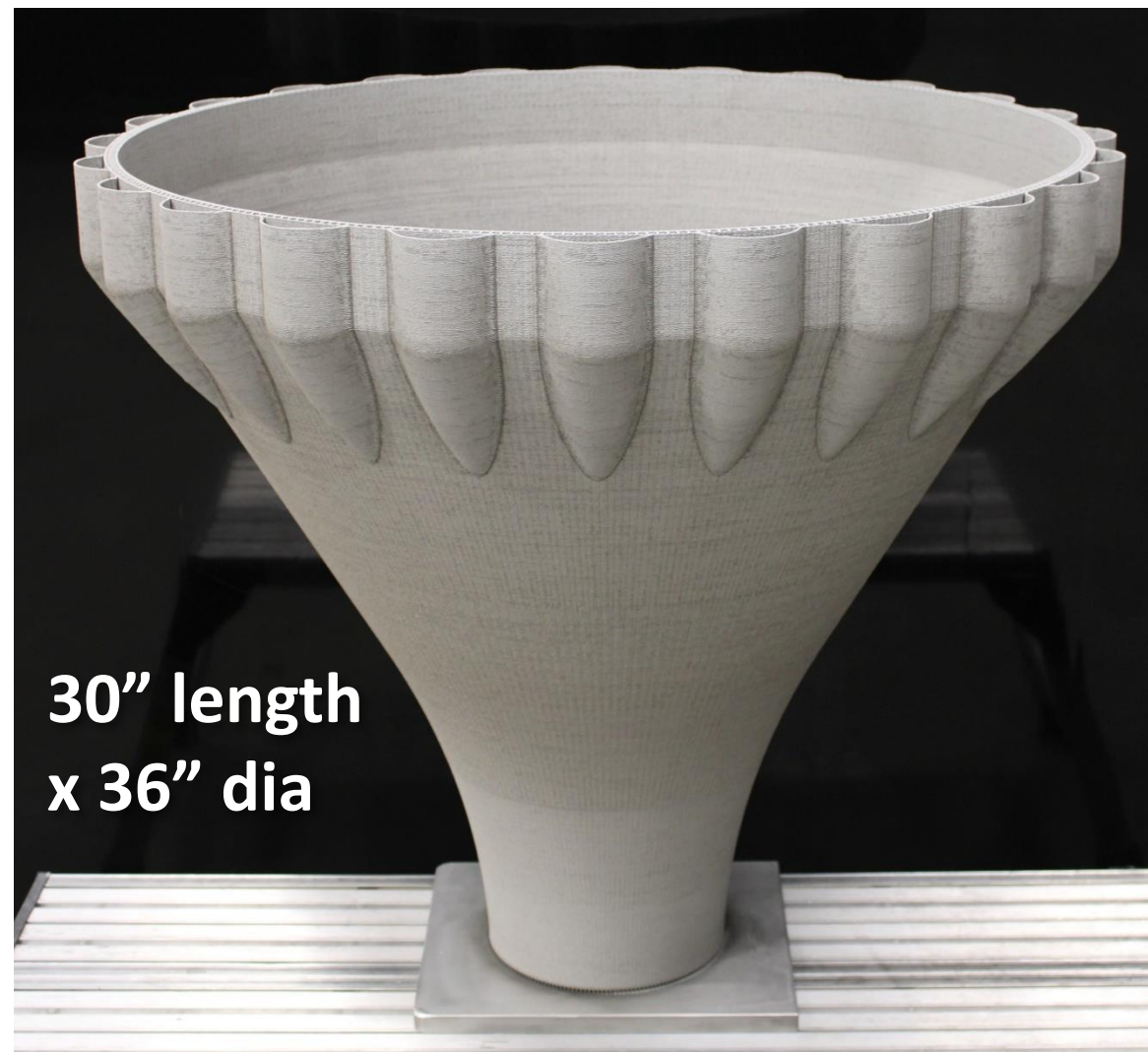
**95" (2.41 m) dia and 111" (2.82 m) height**  
Near Net Shape Forging Replacement



# Demonstrator Aerospike LP-DED Nozzle

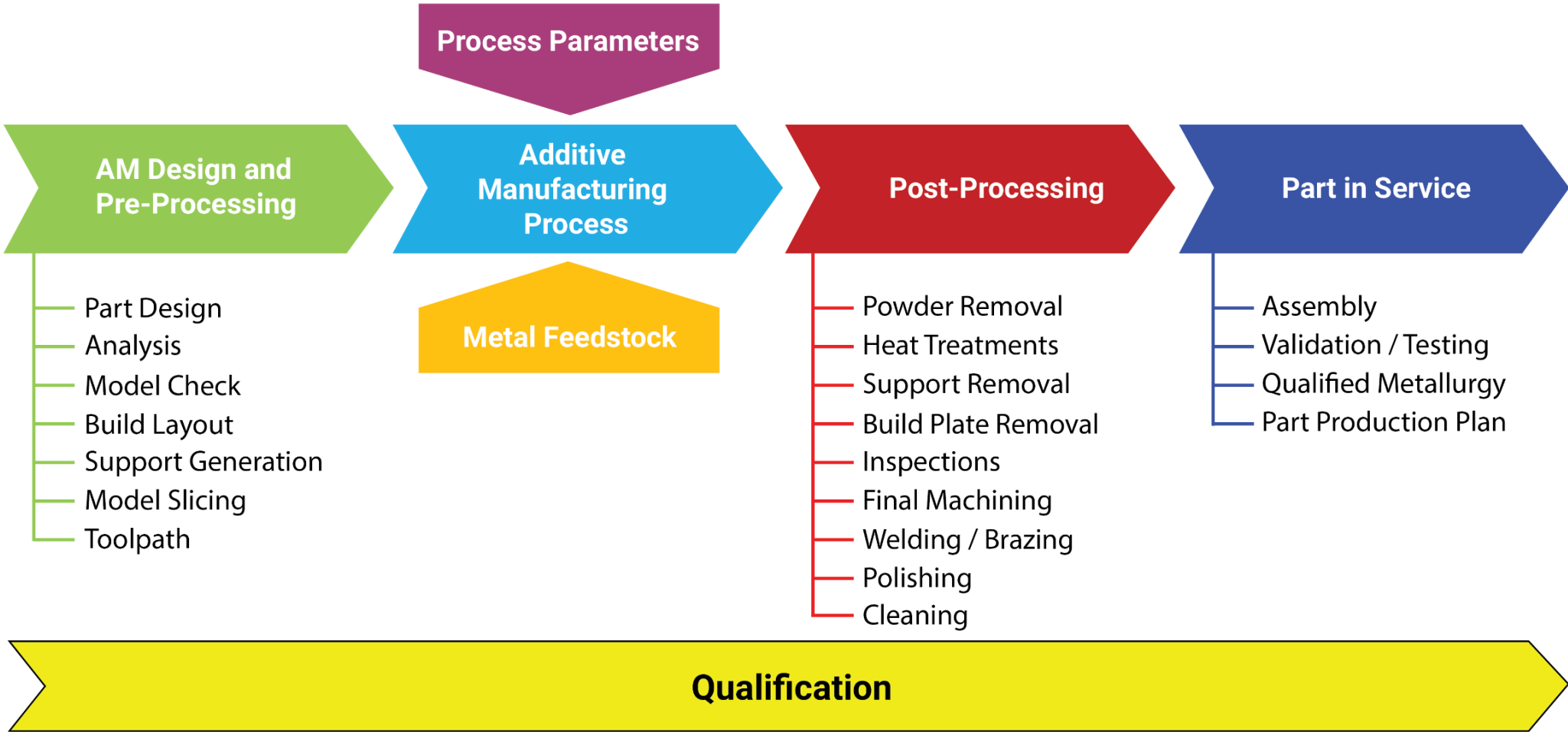


LP-DED Aluminum 6061-RAM2





# Additive Manufacturing Typical Process Flow



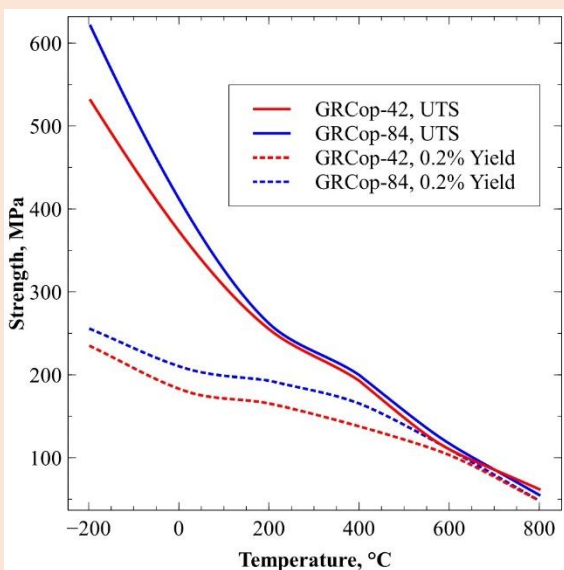
**Proper AM process selection requires an integrated evaluation of all process lifecycle steps**



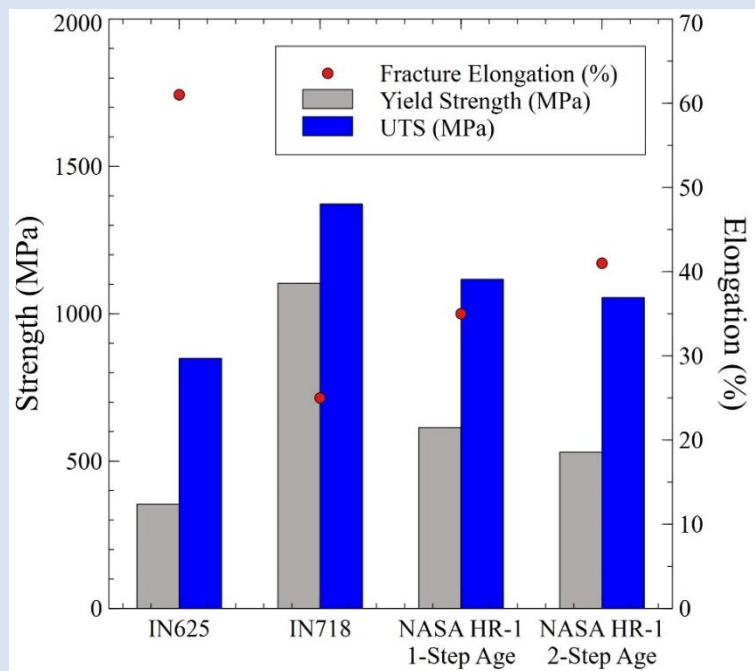
# AM Enabling New Alloy Development



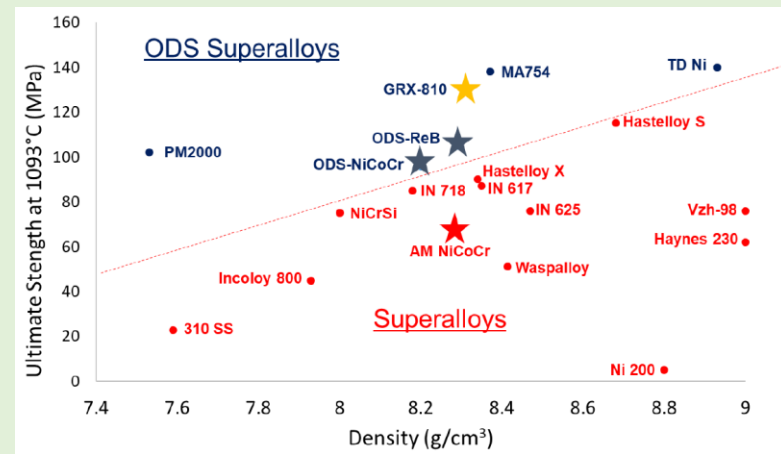
**GRCo-42**, High conductivity and strength for high heat flux applications



**NASA HR-1**, high strength superalloy for hydrogen environments

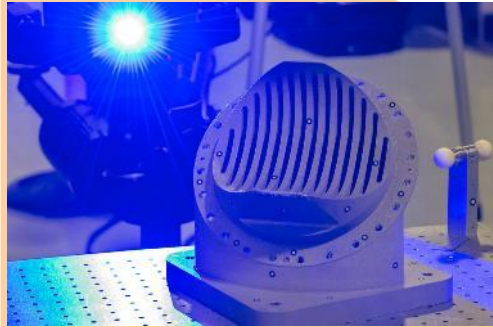


**GRX-810**, high strength, low creep rupture and oxidation at extreme temperatures





# Industrial Maturity and TRL of AM Processes



**L-PBF**

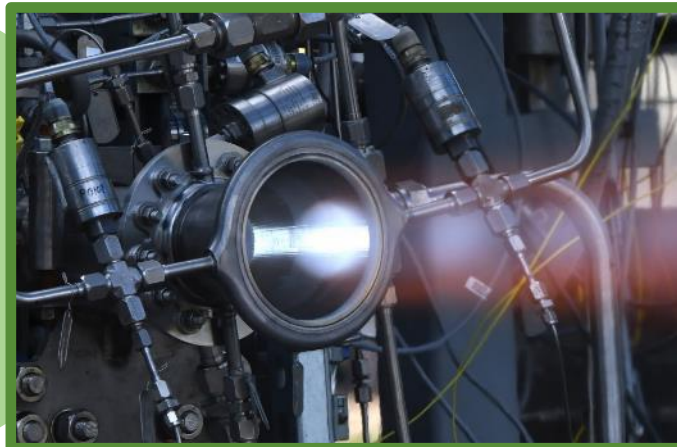


**Cold spray**

**LP-DED**

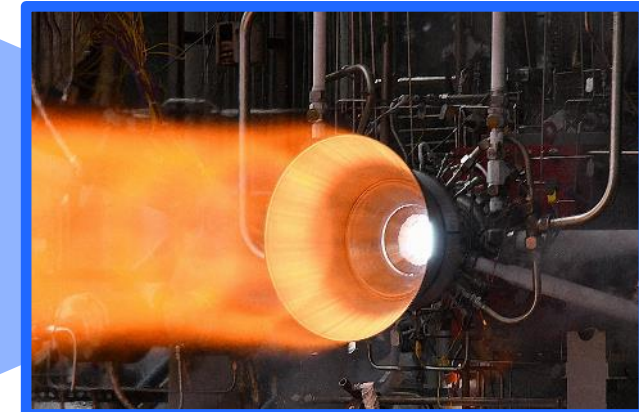


**L-PBF**



**L-PBF**

**EBW-DED**



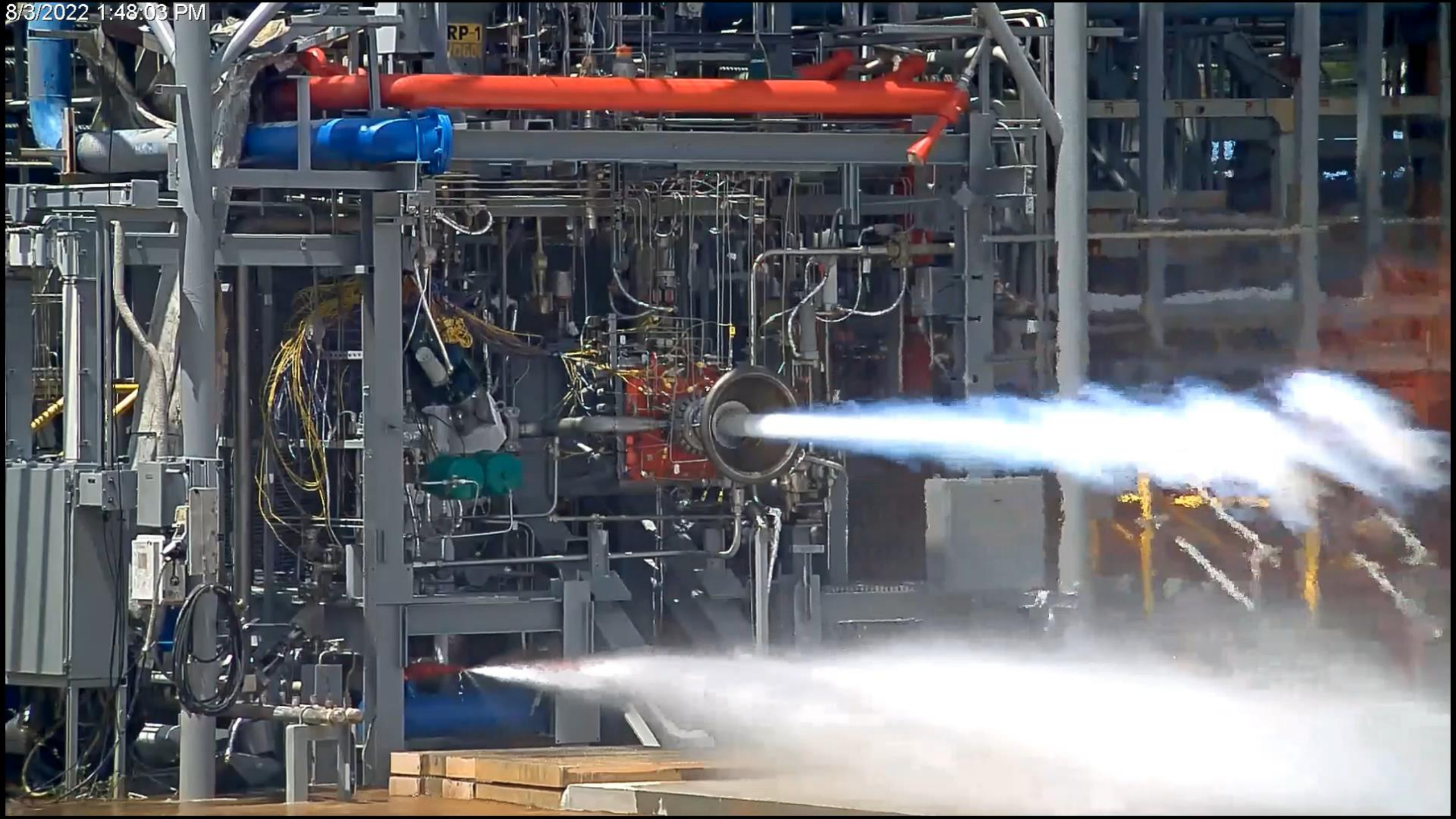
**AW-DED**



**LW-DED**



8/3/2022 1:48:03 PM



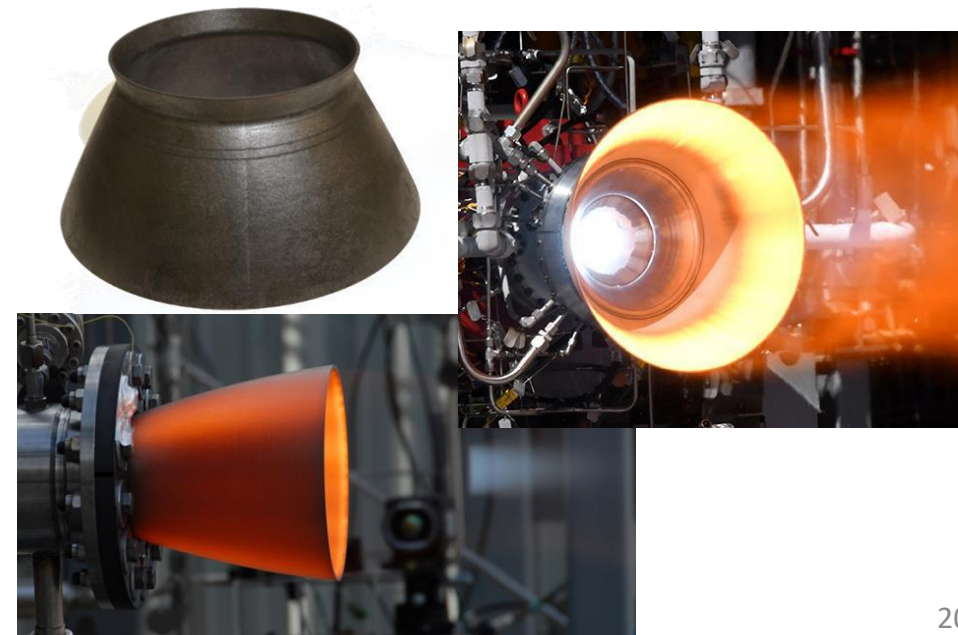




# Carbon-Carbon (C-C) and Carbon Matrix Composite (CMC) High Temp Nozzles

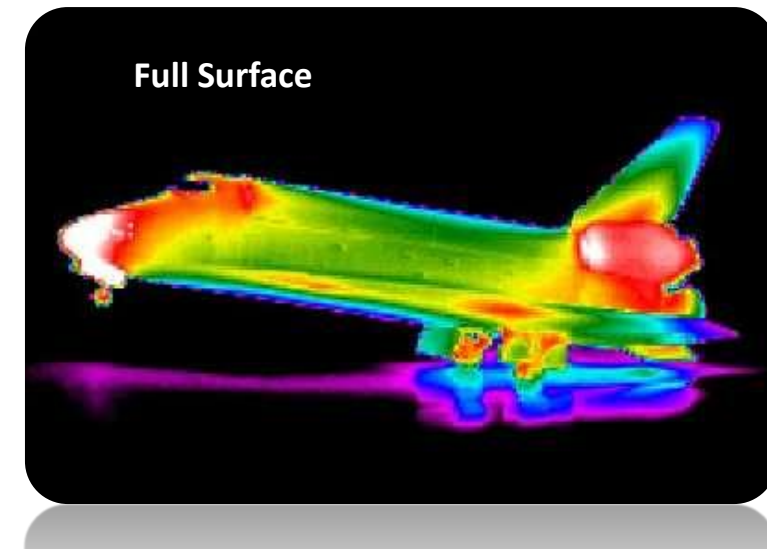
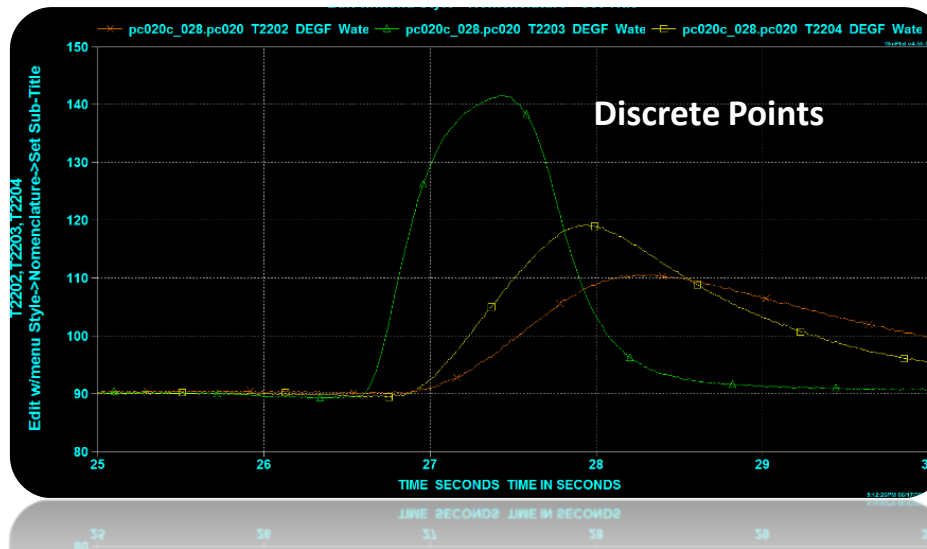


- CNE characterization and material property database development using tag-end rings
  - Nondestructive characterization at MSFC: infrared thermography (IRT) & computed tomography (CT)
  - Mechanical & thermal testing at Southern Research: material properties, sub-element hoop tension tests
  - Tag-ends taken from CNE's for 1.2K-lbf and 35K-lbf engine tests. Provided by: Allcomp, C-CAT, & NGIS
- Materials screening via 1.2K-lbf LOX/GH2, LOX/LCH4, LOX/RP-1 engine testing
  - Investigating coatings, composite matrix chemistry, & attachment concepts
  - Nozzle extensions supplied by: Allcomp, C-CAT, NGIS, PSI
- Mid-scale demonstration via META4X4 7K-lbf LOX/LCH4
  - Completed for lander-class engines
- Large-scale demonstration via 35K-lbf LOX/LH2
  - Performed two low-budget feasibility/demonstration
  - Test series with PAN- & Iyocell-based C-C





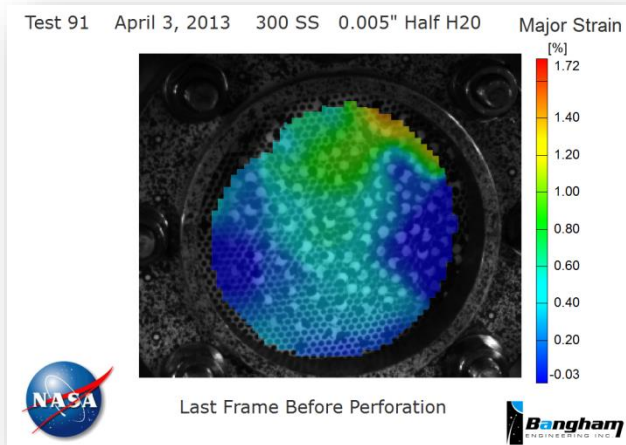
- Subscale and Full-scale testing requires expensive and labor intensive instrumentation to better understand hardware performance
  - Design Modifications and Performance Predictions based on “discrete” point instrumentation
    - Thermocouples, Pressure Transducers, Accelerometers, Strain Gages
- **Challenge: Measure highly dynamic elevated temperature components**



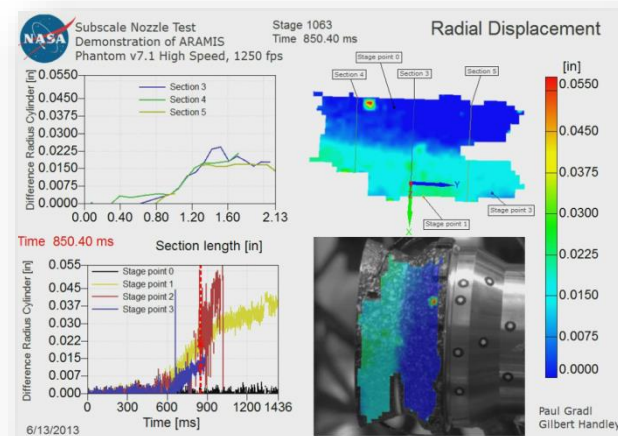
**Goal: Augment Traditional Gages to gain a better understanding of hardware and environment loads to design more efficient components and systems**



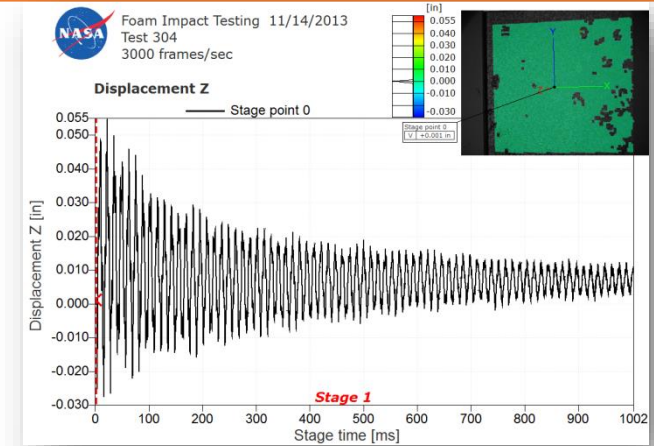
# Applications of Digital Image Correlation (DIC) for Rocket Applications



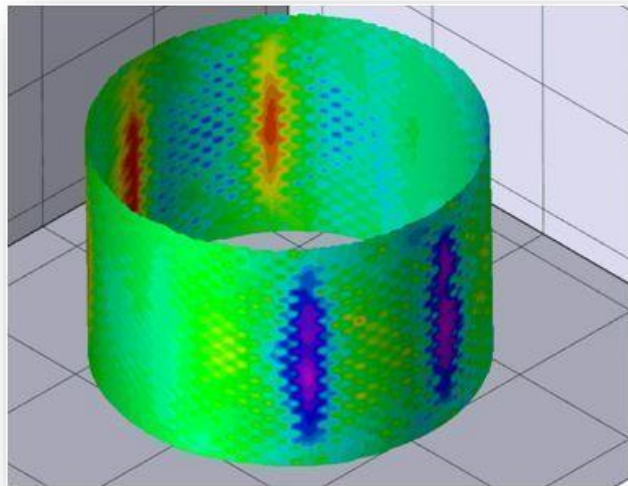
Blast Pressure Wave Tracking at 70,000 fps



Subscale Nozzle Displacements at 1700F

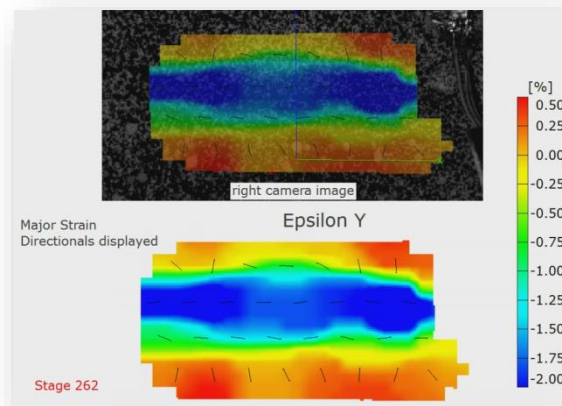


Debris Impact Testing –  
Eliminated Strain Gages



Full-Field Strain and Displacements  
of 18-ft Dia Tank

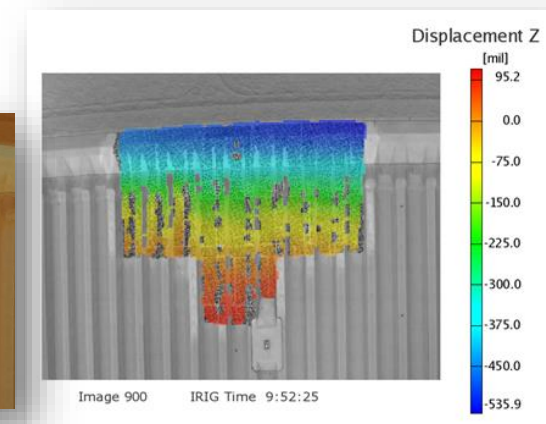
Ref: Todd Boles, MSFC/ET30



High Speed Composite Compression  
– Direct Application of Major Strain



ET (on Pad) Cryo tanking test to observe  
stringer displacement



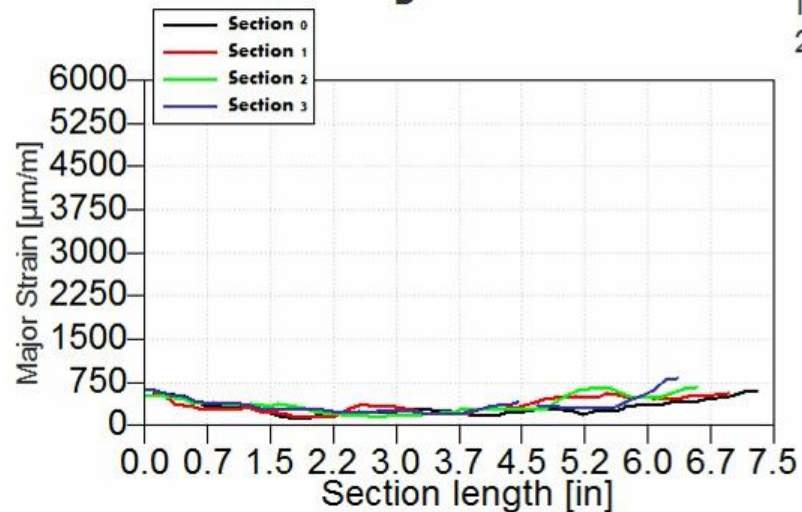


# Tracking Real-time strain on hardware



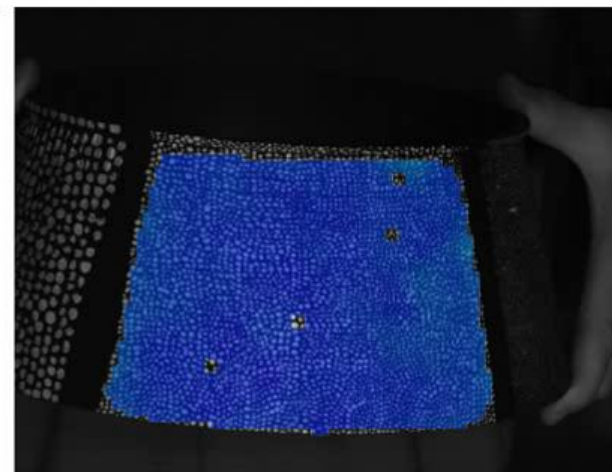
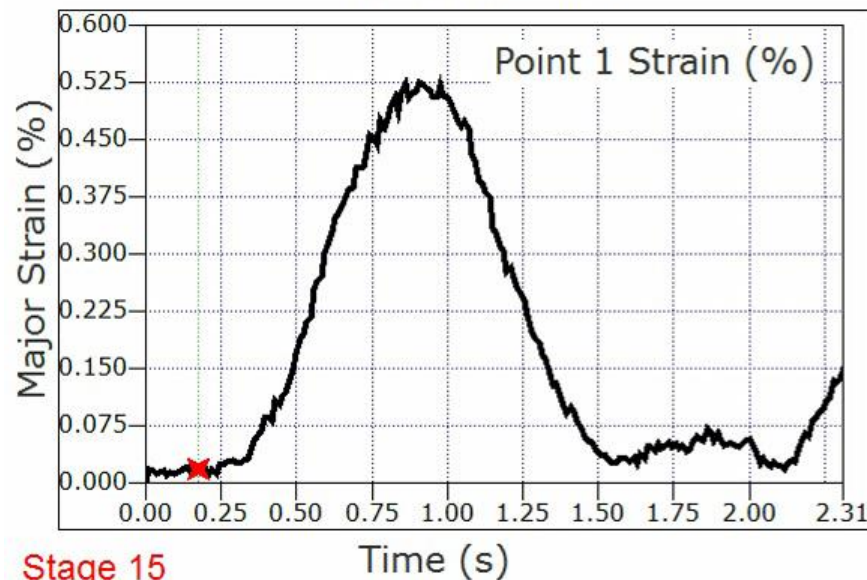
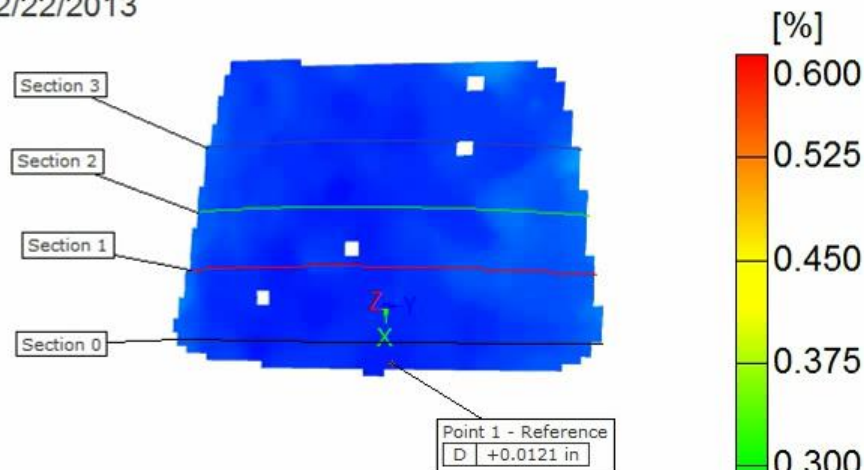
Stage 15

## Nozzle - Major Strain



Stage 15  
Time 0.17 s  
2/22/2013

## Major Strain



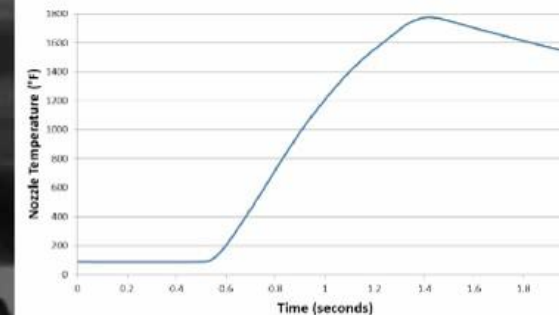
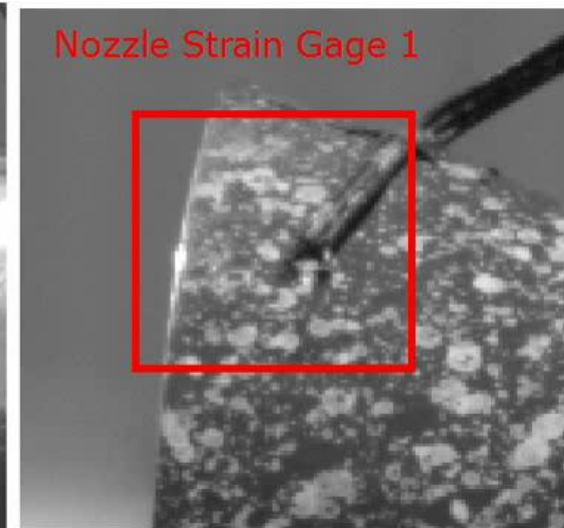
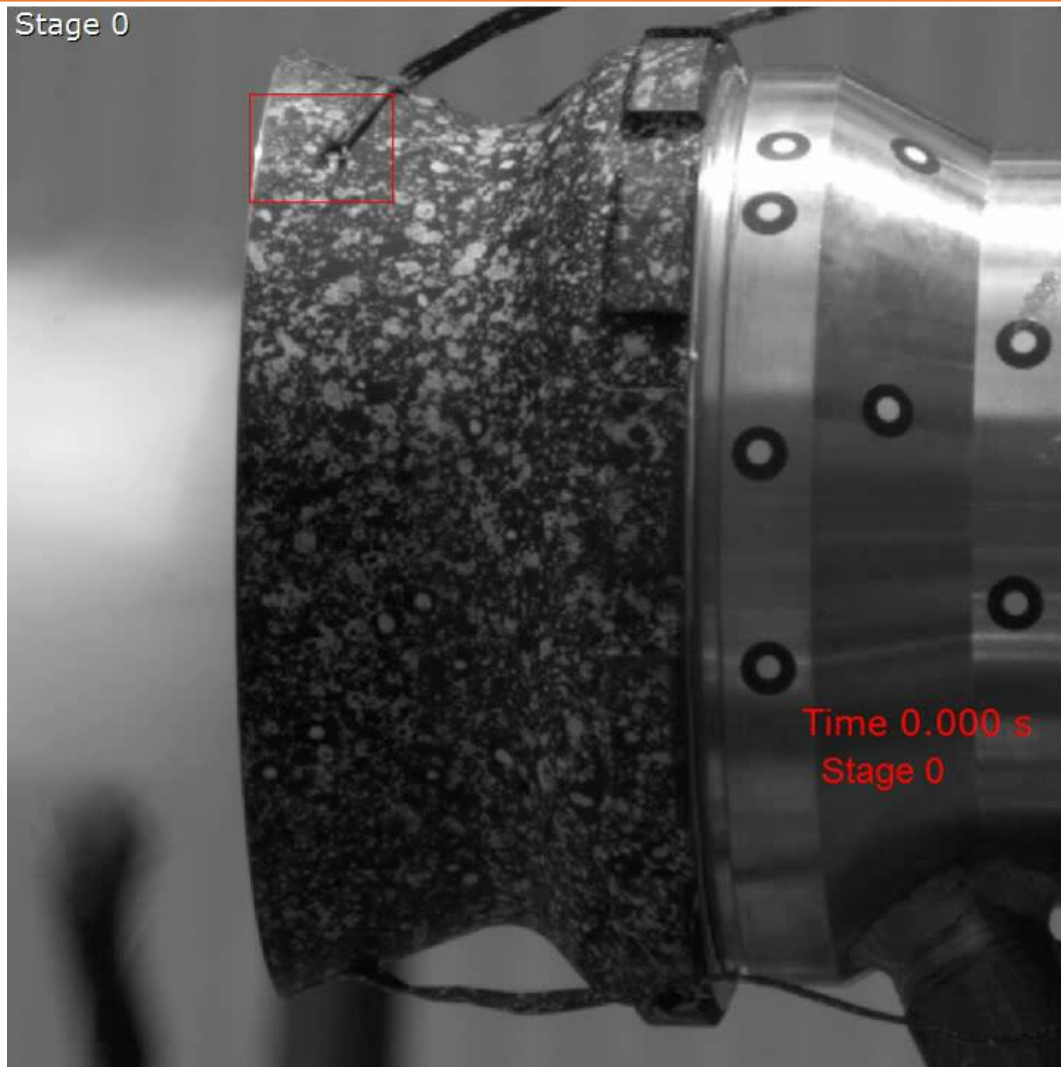
Stage 15

Paul Gradl  
Gilbert Handley

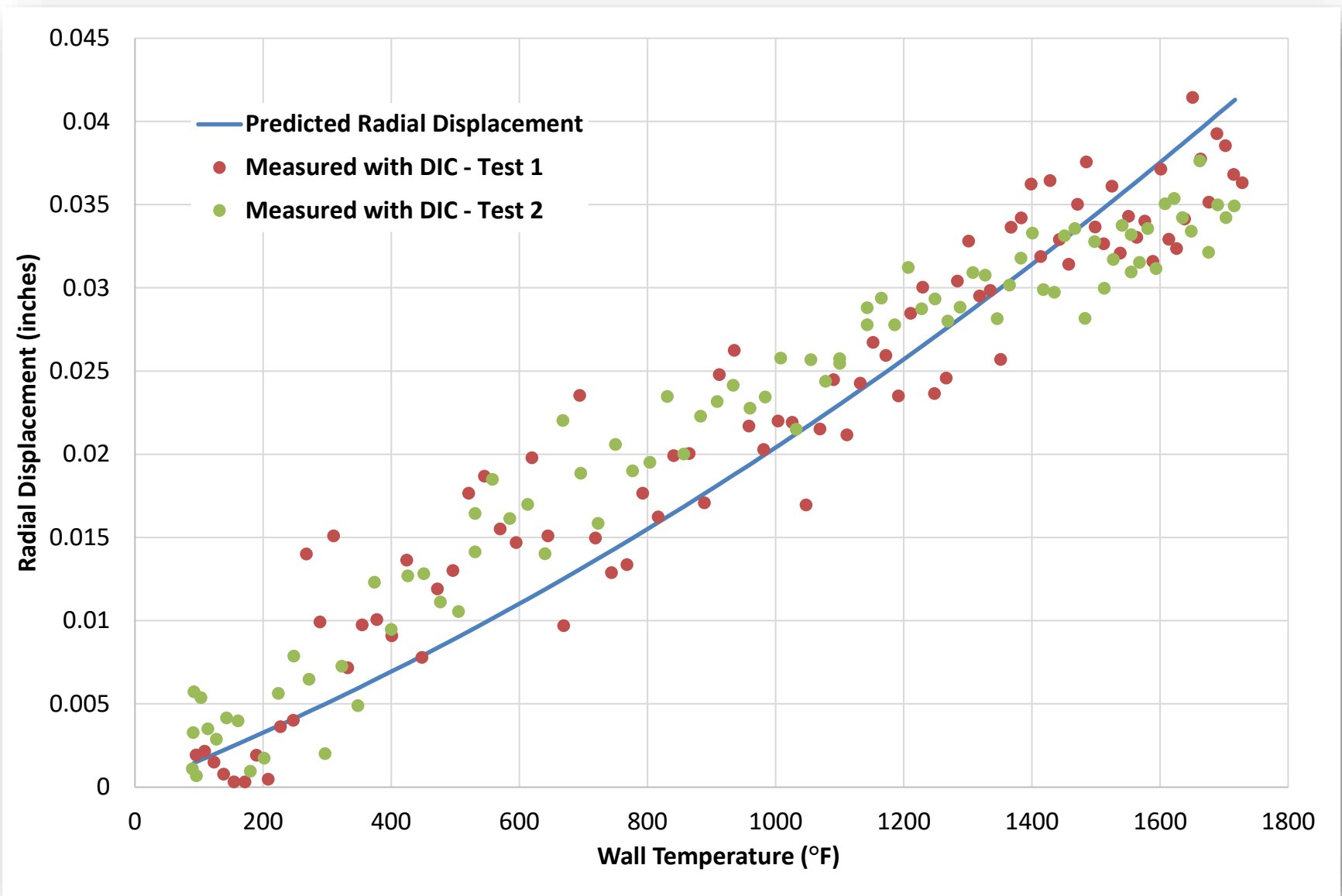




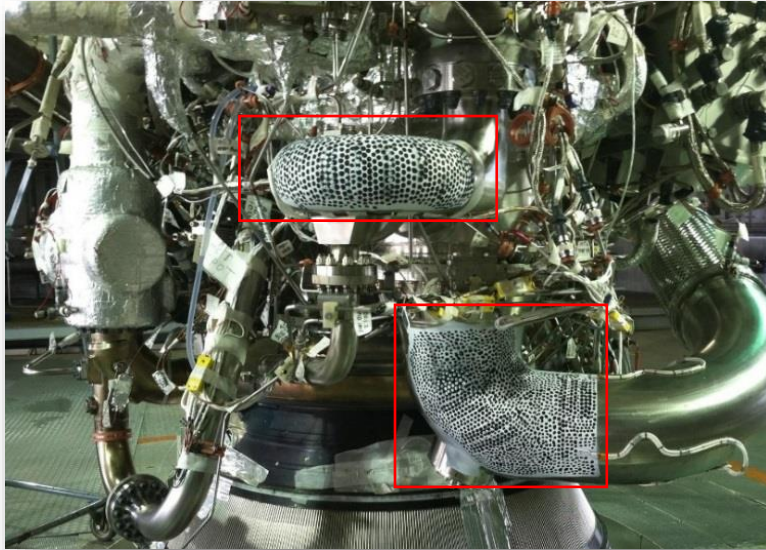
# Tracking Strain using DIC at high temperatures on rocket nozzles



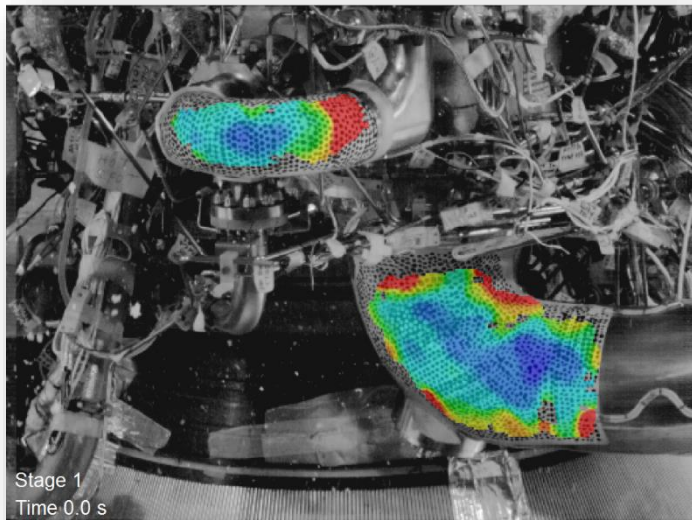
# Subscale Hotfire Testing – Data Analysis







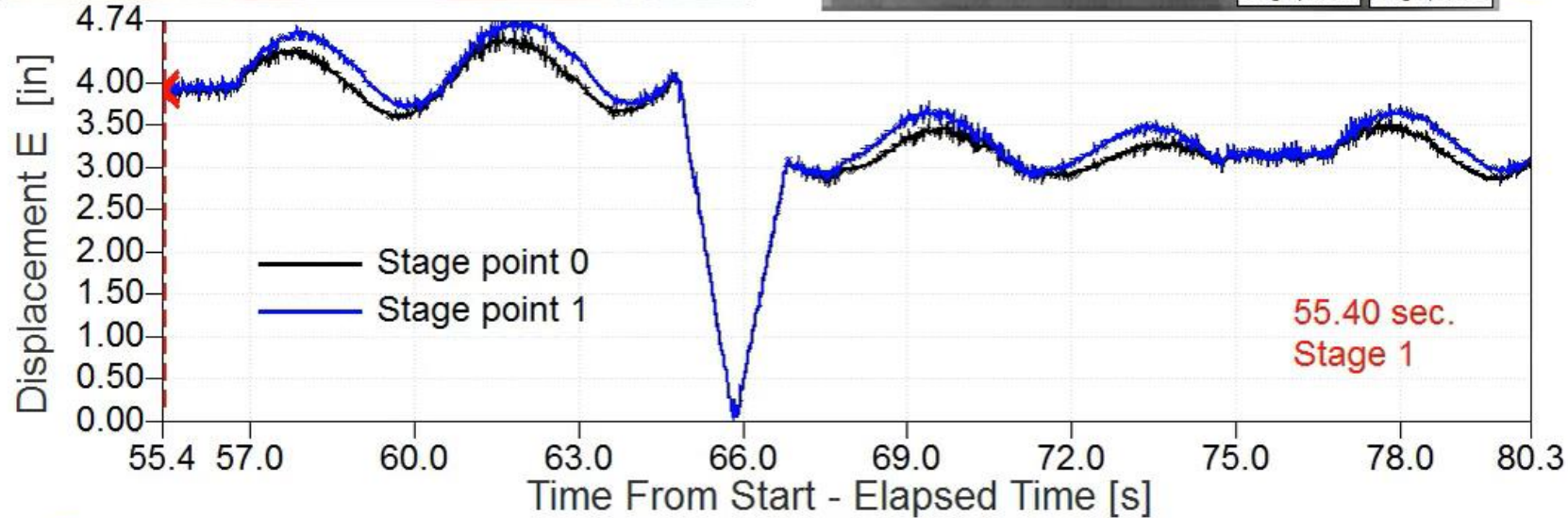
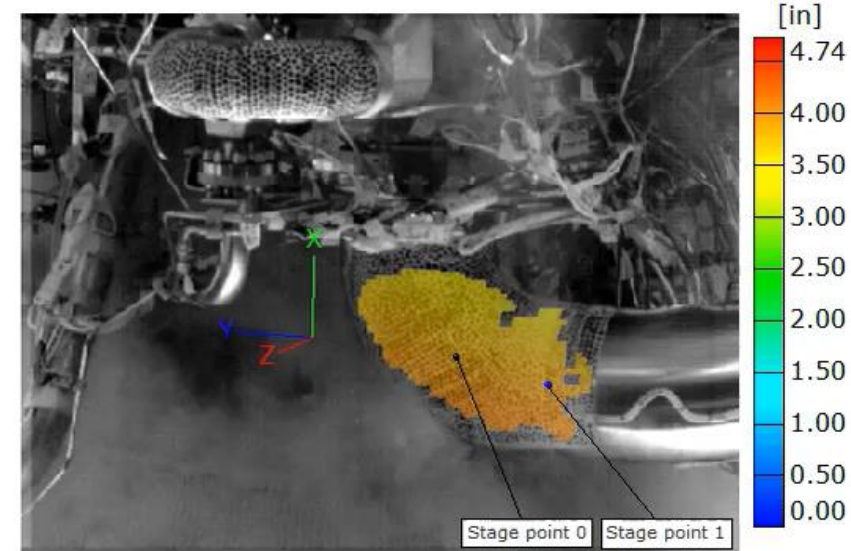
**Stereo Cameras installed and Speckle Pattern Applied at Stennis A1 Stand**



**High speed cameras for DIC**



# Results from Full Scale Engine Testing



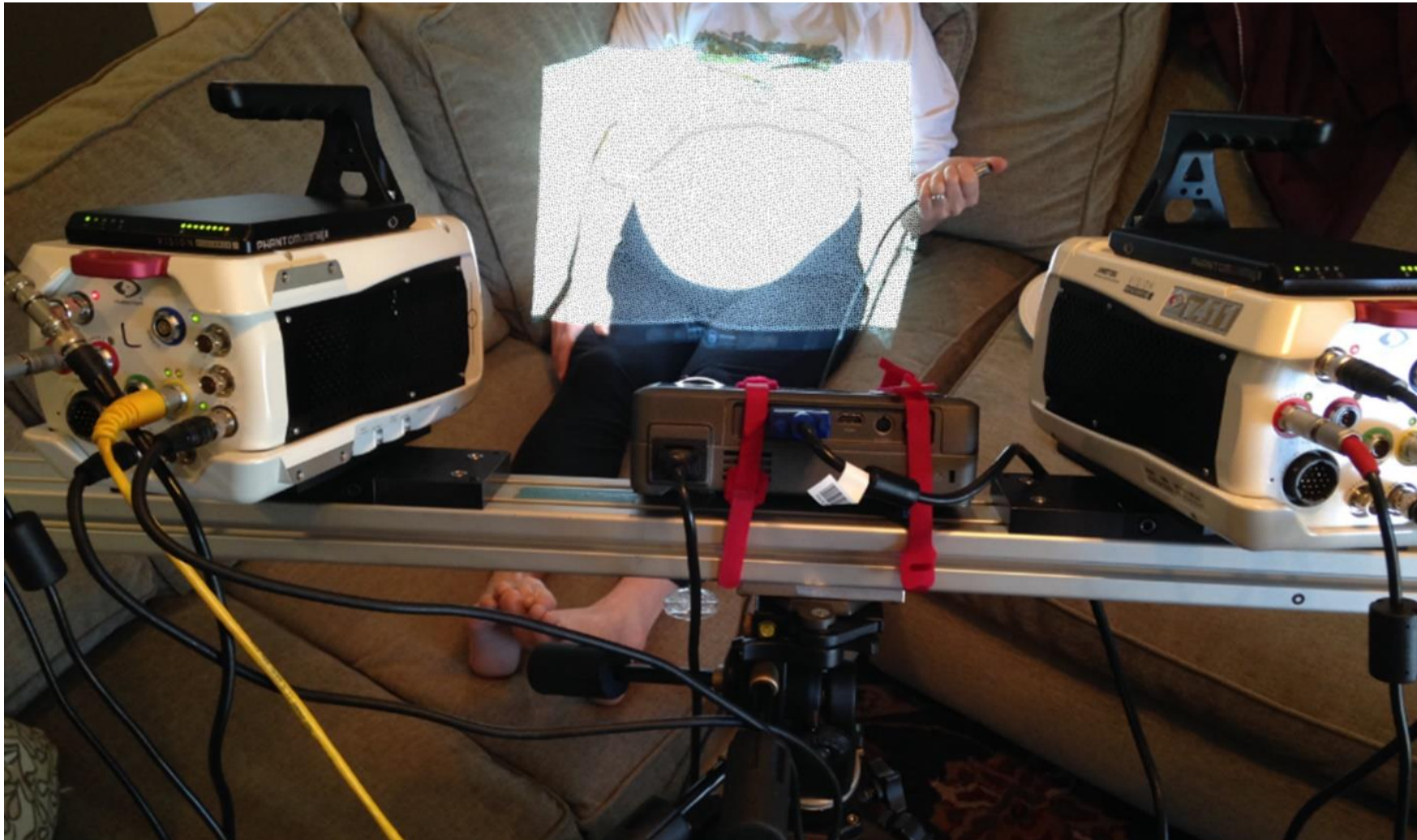


Dynamic responses  
require an input to  
excite the system...

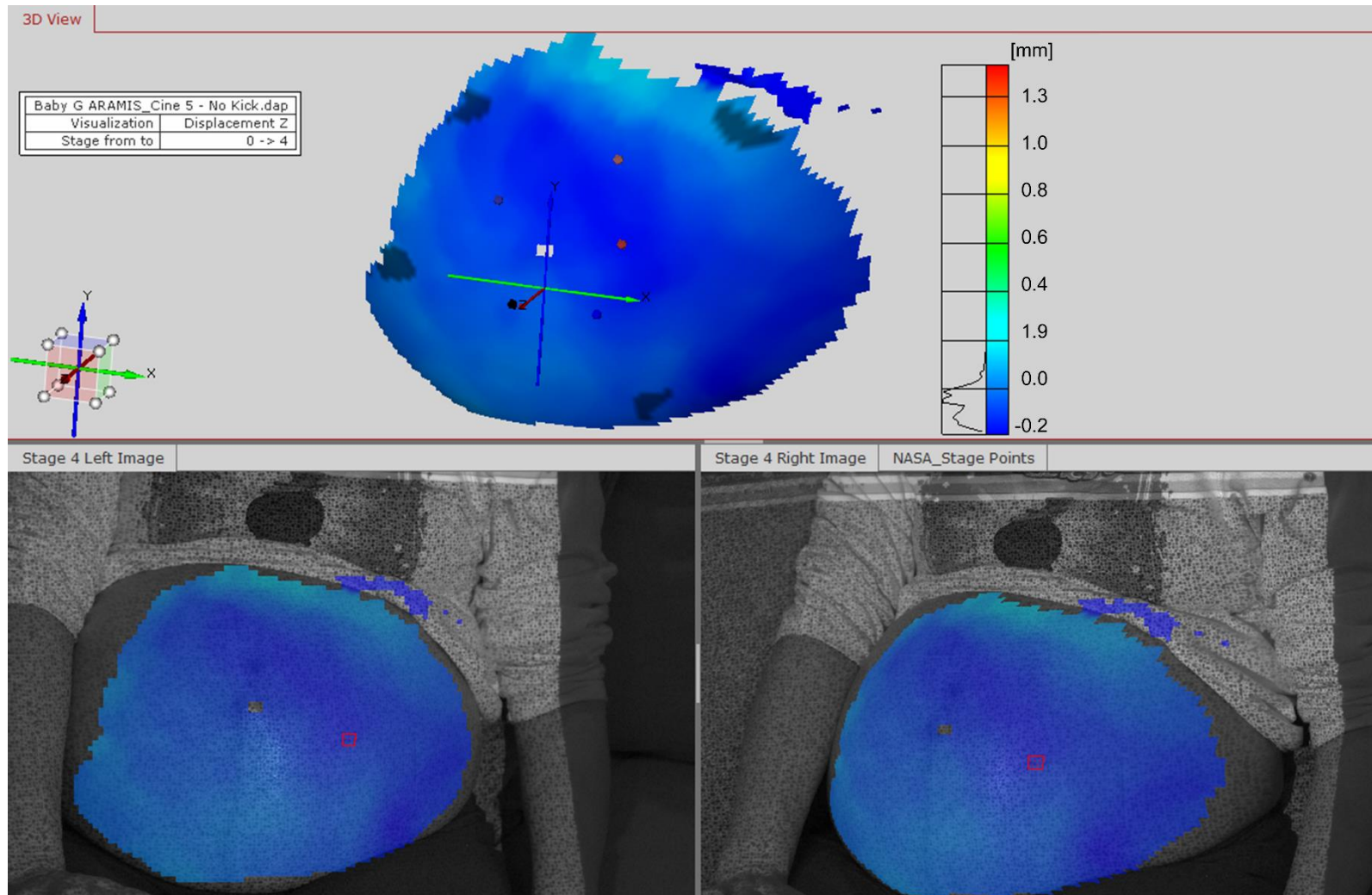


# Non-contact Optical Measurements

Images were collected using a projected pattern instead of painting a speckle pattern on her belly... High Speed cameras were post triggered after movements felt.

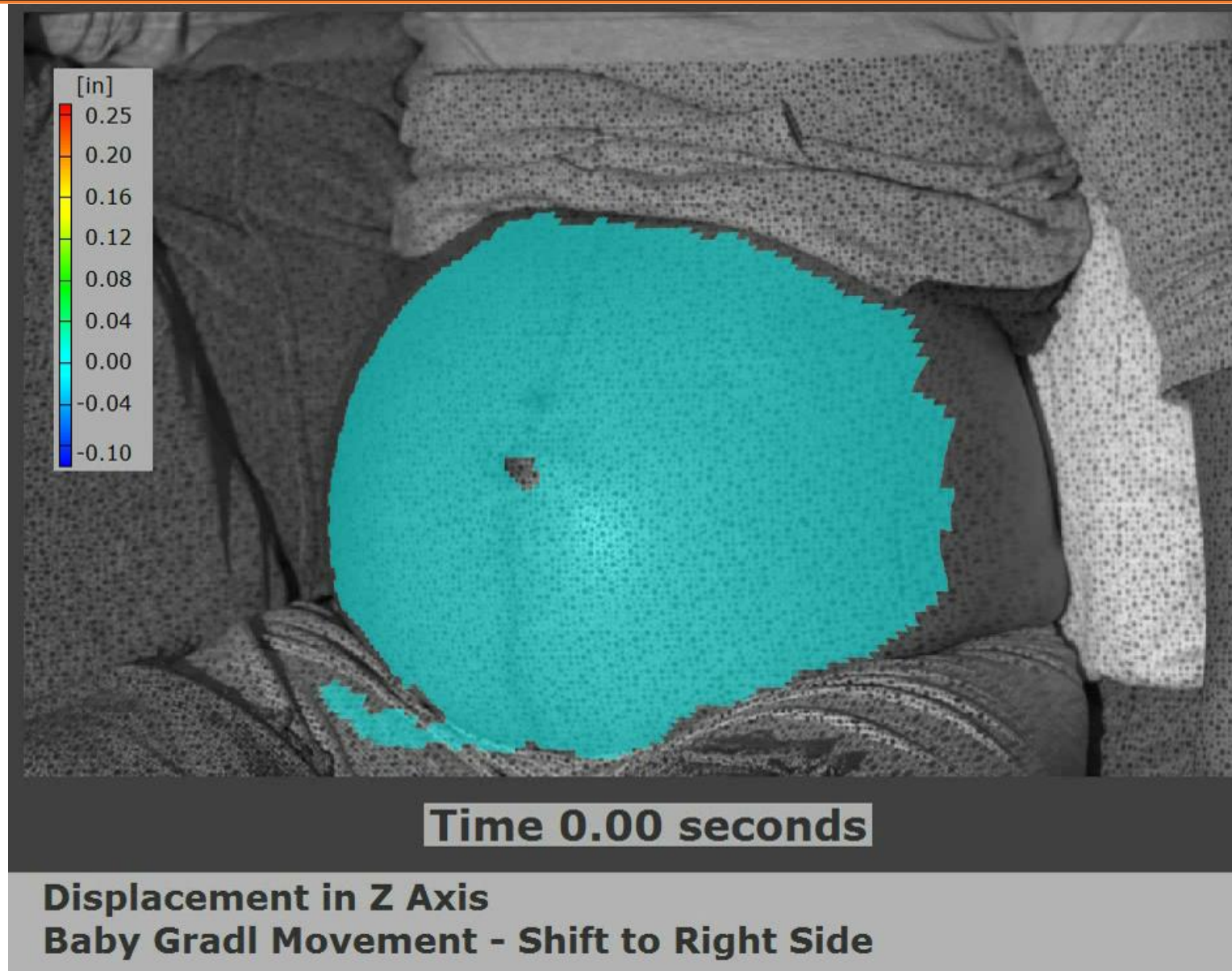


To ensure that kicks and movement data was real a background test was conducted with no baby movement (to correct for breathing and body motion)

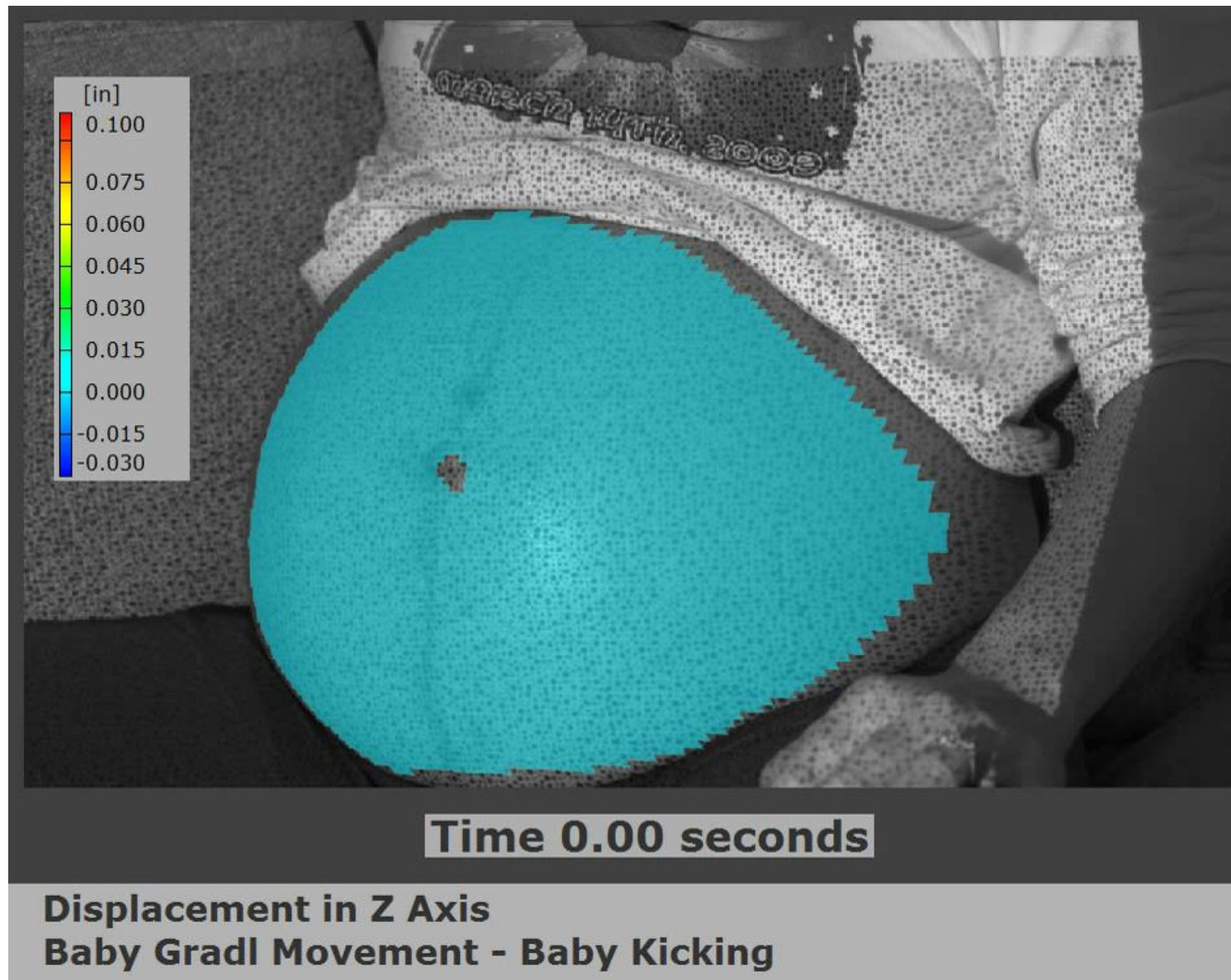




# Non-contact Optical Measurements

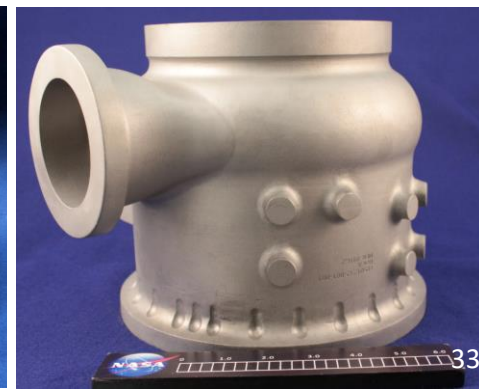
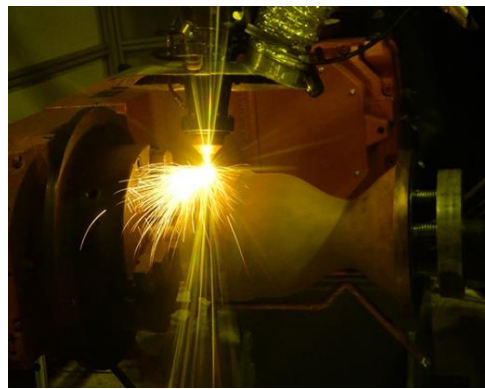
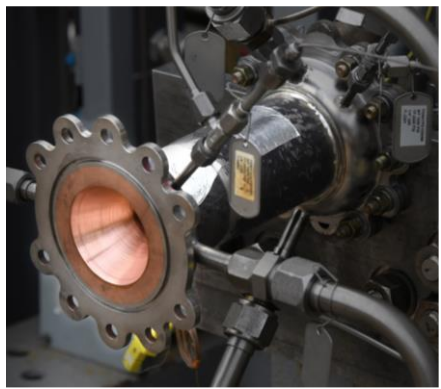


# Non-contact Optical Measurements

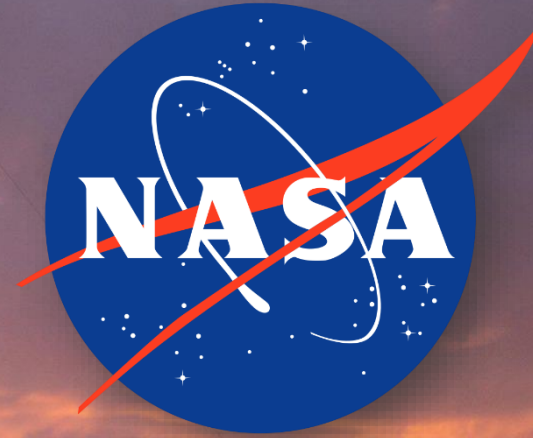




- Various advanced manufacturing and AM processes have matured for rocket propulsion applications each with unique advantages and disadvantages.
- AM is not a solve-all; consider trading with other manufacturing technologies and use only when it makes sense.
- **Complete understanding of the design process, build-process, feedstock, and post-processing is critical to take full advantage of AM.**
- Additive manufacturing takes practice!
- Standards and certification of the AM processes are in-work.
- AM is evolving and imagination is the limit.







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# NASA led book on Metal Additive Manufacturing



## Metal Additive Manufacturing for Propulsion Applications

Edited by  
Paul R. Gradl, Omar R. Mireles,  
Christopher S. Protz, and Chance P. Garcia



PROGRESS IN ASTRONAUTICS AND AERONAUTICS

Timothy C. Liewen, Editor-in-Chief  
Volume 263

<https://arc.aiaa.org/doi/book/10.2514/4.106279>

Online version and hardcopy available

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<https://arc.aiaa.org/doi/book/10.2514/4.106279>

Additive manufacturing (AM) processes are proving to be a disruptive technology and are grabbing the attention of the propulsion industry. AM-related advancements in new industries, supply chains, design opportunities, and novel materials are increasing at a rapid pace. The goal of this text is to provide an overview of the practical concept-to-utilization lifecycle in AM for propulsion applications.